# FIBER-OPTIC INSTRUMENTATION FINAL REPORT TECHNICAL REPORT A001:

NASA CR-160448

# CRYOGENIC SENSOR MODEL DESCRIPTION

CONTRACT NO. NAS9-15454

**25 NOVEMBER 1979** 

PREPARED FOR

NASA/JSC HOUSTON, TEXAS

(NASA-CR-160448) FIBER-OPTIC
INSTRUMENTATION: CRYOGENIC SENSOR MODEL
DESCRIPTION Final Report (TRW Defense and
Space Systems Group) 20 p HC A02/MF A01

N80-17848

A01 Unclas CSCL 20F G3/74 47115





# FIBER-OPTIC INSTRUMENTATION FINAL REPORT **TECHNICAL REPORT A001:** CRYOGENIC SENSOR MODEL DESCRIPTION

CONTRACT NO. NAS9-15454

**25 NOVEMBER 1979** 

PREPARED FOR

NASA/JSC HOUSTON, TEXAS

P	R	F	P	Δ	R	F	n	R	<b>Y</b> :	
•		_	•	_		_	u	_		

They are Horas warns M.M. SHARMA PROJECT MANAGER,

FIBER-OPTICS INSTRUMENTATION

**APPROVED BY:** 

P. STAUDHAMMER MANAGER,

ADVANCED TECHNOLOGY LABORATORY



# CONTENTS

			Page
1.	INTR	RODUCTION AND SUMMARY	1-1
	1.1	Background and Purpose of Study	1-1
	1.2	Summary of Study and Recommended Sensor Conce	epts 1-2
	1.3	Objectives	1-3
	1.4	Conclusions	1-3
	1.5	Recommendations	1-4
		1.5.1 Liquid-Level Sensor 1.5.2 Pressure Gauge 1.5.3 Temperature Sensor	1-4 1-4 1-5
2.	SENS	SOR SELECTION	2-1
	2.1	Specific Concepts Considered	2-1
		<ul><li>2.1.1 Liquid-Level Sensor</li><li>2.1.2 Temperature Sensor</li><li>2.1.3 Pressure Sensor</li></ul>	2-1 2-5 2-5
	2.2	Overview of Study Results	2-15
	2.3	Concepts Selected	2-15
		<ul><li>2.3.1 Liquid-Level Sensor</li><li>2.3.2 Pressure Sensor</li><li>2.3.3 Temperature Sensor</li></ul>	2-15 2-21 2-21
3.	EXPE	ERIMENTAL MODEL	3-1
	3.1	Performance Requirements	3-1
		3.1.1 Environment 3.1.2 Operation Mode 3.1.3 Level Accuracy 3.1.4 Temperature Range 3.1.5 Pressure Range 3.1.6 Time Response 3.1.7 Sensor Installation Accuracy 3.1.8 Sensor Location 3.1.9 Flow Location 3.1.10 Sensor Lifetime 3.1.11 Lifetime of Electronics	3-4 3-4 3-4 3-4 3-5 3-5 3-5 3-6 3-6
	3.2	Test Procedures	3-6
		<ul><li>3.2.1 Initial Setup</li><li>3.2.2 Cool-Down and Moisture-Purging Procedures</li><li>3.2.3 Calibration Procedures</li><li>3.2.4 Demonstration Procedure</li></ul>	3-7 3-7 3-10 3-12
	3.3	Electronic Subsystem	3-12

# CONTENTS (Continued)

		Page
4.	SENSOR DEMONSTRATION MODEL DESIGN DESCRIPTION	4-1
	4.1 Liquid-Level Sensor	4-1
	4.2 Pressure Gauge	4-4
	4.3 Temperature Gauge	4-8
	REFERENCES	R-1
	APPENDIY A	A_1

# ILLUSTRATIONS

		Page
2-1	Recommended Liquid-Level Point Sensor Concept	2-4
2-2	Recommended Temperature-Sensor Concept	2-8
2-3	Temperature Dependence of Absorption Bandedge Location for GaAs	2-9
2-4(a)	Instrument Concept for Semiconductor Bandedge Temperature Sensor: Reference Temperature Cycling	2-10
2-4(b)	Instrument Concept for Semiconductor Bandedge Temperature Sensor: Spectral Analysis	2-11
2-4(c)	Instrument Concept for Semiconductor Bandedge Temperature Sensor: Ratio of 2-Color Outputs	2-12
2-5	Summary of Recommended Sensor and Instrument Concepts	2-16
2-6	Selected Fluid-Level Sensor Concept	2-18
2-7	Schematic of Light Coupling Between Input and Output Fibers	2-19
2-8	Schematic of TRW Level-Sensing System	2-22
2-9	Recommended Pressure-Sensor Configuration	2-23
2-10	Electronic Circuit Associated with Pressure Sensor	2-24
2-11	Temperature-Tuned LED Output Sweeping Through Absorption Edge of Sensor Crystal	2-25
2-12	Data Bandedge of Indium Phosphide and Gallium Arsenide vs. Temperature	2-27
2-13	Schematic of Temperature-Sensing System and Associated Input/Output Waveforms	2-28
3-1	Experimental Model	3-2
3-2	Sting Assembly	3-3
3-3	Experimental-Model Fiber-Optic Connections	3-8
3-4	Oscilloscope Connections for Experimental-Model Electronics Calibration	3-11
i <b>-</b> 5	Block Diagram of Experimental-Model Electronic Subsystem	3-11

# ILLUSTRATIONS (Continued)

		Page
3-6	Block Diagrams for Liquid-Level, Pressure, and Temperature Gauges	
3-7		3-14
	Block Diagram of Temperature-Sensor Resolving Electronics	3-16
4-1	Final Liquid-Level Sensor	4-3
4-2	Level-Gauge Electronic Schematic	4-6
4-3	Diaphragm Displacement vs. Reflected Light	4-7
4-4	Pressure-Gauge Construction Details	4-10
4-5	Pressure-Gauge Electronic Schematic	4-11
4-6	Schematic of Master Oscillator	4-12
4-7	Schematic of LED Drivers with Feedback Control	4-12
4-8	Bandedge Curve of GaAs Crystal	
4-9	Oscillograms Showing Bandedge vs. Temperature for GaAs	4-14
		4-15
<b>1</b> -10	Block Diagram of Temperature-Gauge Electronics	4-17
1-11	Temperature-Gauge Mechanical Arrangement	4-19
-12	Temperature-Gauge Electronic Schematic	
-13	25-Amp Laser-Diode Pulser Electronic Schematic	4-20
	Stode ruiser Electronic Schematic	4-21

# TABLES

		Page
2-1	Liquid-Level Sensor Concepts Considered	2-2
2-2	Evaluations of Point Level-Sensor Concepts	2-3
2-3	Temperature-Sensor Concepts Considered	2-6
2-4	Evaluation of Temperature-Sensor Concepts	2-7
2-5	Pressure-Sensor Concepts Considered	2-13
2-6	Evaluation of Pressure-Sensor Concepts	2-14
2-7	Sensor Scoring Evaluation	2-17
4-1	Liquid-Level Sensors Insertion-Loss and Sensitivity Data	4-2
4-2	Signal Loss Due to Epoxy	4-5
4-3	Pressure-Gauge Calibration Readings for Gauge Operating in ${\rm LN}_2$	4-9
4-4	Pressure-Gauge Lookup Table for Gauge Operating in LN <sub>2</sub>	4-9
4-5	Temperature-Gauge Calibration Readings for Gauge Operating in ${\sf LN}_2$	4-18
4-6	Temperature-Gauge Lookup Table for Gauge Operating in LN2	4-18

# 1. INTRODUCTION AND SUMMARY

The development of fiber-optic systems and of passive optical sensors appropriate for measurement of conditions in cryogenic liquid propellant tanks in accordance with requirements of NASA/JSC Contract NAS9-15454 entitled "Fiber-Optic Instrumentation" has been completed. The purpose of this report is to outline the overall program undertaken by TRW, to document the research, development, and testing performed under this contract, and to identify recommendations for subsequent development programs.

# 1.1 BACKGROUND AND PURPOSE OF STUDY

United States space programs use launch-vehicle engines requiring highly explosive propellants. For example, the Space Shuttle, the current launch vehicle of the space program, incorporates main engines that are fueled by liquid oxygen and liquid hydrogen – both of which are extremely explosive liquids. Every effort is made to reduce the potential hazards associated with handling and using these propellants. However, a large portion of the cost of using hazardous materials such as LOX is reflected in the necessary safety measures that must be implemented. Therefore, a small improvement in increasing safety results in a large reduction in cost.

In the case of the Space-Shuttle  $LH_2$  and LOX tanks, the instrumentation that performs liquid-level gauging principally consists of point electrical sensors, which are a proven and common method for gauging tanks. Capacitance gauges are also employed, particularly when a continuous reading is required. Although from the engineering standpoint point sensors and capacitance guages can be made to work well in a LOX or  $LH_2$  tank, from the safety standpoint, both techniques are undesirable because the power and signal leads to the sensor must penetrate the tank, thus presenting the possibility of an electrical anomaly capable of causing an explosion.

When a hazardous condition exists, such as that of point sensors within LOX tanks, the risk has been controlled by very careful quality control and extensive testing to reduce the possibility of a failure to an extremely low number. The result of this special handling and additional record keeping has been imposition of a significant increase in the overall cost of inexpensive hardware. Obviously, if current-carrying wires can be eliminated from tanks that contain material susceptible to ignition, then both increased safety and lower costs are realized.

Recent advances in fiber-optic technology are leading to wider application of such systems, including use in passive optical-sensing devices. The purpose of present TRW programs in fiber-optic instrumentation has been to develop applicable technology to replace current-carrying signal and power wires of point-sensors, temperature sensors, and presure sensors with fiber-optic systems.

# 1.2 SUMMARY OF STUDY AND RECOMMENDED SENSOR CONCEPTS

In the Concept Study, general sensor concepts initially were examined to classify types of sensors which would be applicable to the cryogenic regime. Subsequently, specific sensor concepts were studied in detail, in some cases with special laboratory measurements made to facilitate choosing of the best concepts. Library and literature surveys were conducted, and concentrated efforts to generate new and novel concepts and/or adaptations of older ones were undertaken.

Final TRW recommendations were based on a scoring procedure devised to reflect the contractual objective of "developing the technology to replace current-carrying signal and power wires of point sensors, temperature, and pressure sensors with fiber optics." Simplicity and reliability were particularly emphasized in this evaluation because sensor operation will occur within the stringent, cryogenic temperature range. Sensor concepts recommended by TRW were the following:

Liquid-Level Point Sensor

Bare-fiber concept used as "ON-OFF" level indication.

• Temperature Sensor

Semiconductor bandedge absorption shift used with instrumentation concept of matching bandedge location with a temperature-cycled reference.

Pressure Sensor

Movement of a pressure-sensitive deflecting membrane sensed via an optical readout device.

Criteria used in evaluating the sensor concepts considered and details of rankings assigned to the various sensors considered are identified in Section 2, selected sensors are described in Section 3, and test results for the completed sensor system are given in Section 4. Principles of fiber-optic systems are summarized in a previous report (Ref. 1).

#### 1.3 OBJECTIVES

Objectives of this investigation were directed towards achieving the following critical goals:

- Assessment and determination of technology requirements for developing a demonstration model to evaluate feasibility of practical cryogenic liquid-level, pressure, and temperature sensors.
- Construction of a demonstration model to measure characteristics of the selected sensor and to develop test procedures.
- Development of an appropriate electronic subsystem to operate the sensors.

These goals were achieved by review of prevailing concepts, matching requirements with available concepts to develop evaluation criteria, and design, development, and test of selected concepts, with modification of basic approaches where necessary.

Because the main consideration during this effort was to design practical gauging systems suitable for use in actual systems, the complete experimental system is housed in a suitable cabinet. The electronic display panel consists of pressure and temperature readouts, indicating lights for the liquid-level sensors, and calibration controls. The liquid nitrogen Dewar is installed on the side of the cabinet.

#### 1.4 CONCLUSIONS

Completed study efforts have led to the following conclusions:

- Available fiber-optic sensor technology pertains to sensing of high-refractive-index liquids at or near room temperature.
- Concepts developed in this program indicate definite feasibility of applying fiber-optic instrumentation to practical usage.
- Development of the demonstration model has identified several areas of beneficial modifications to the present practical designs.

- Because the sensing systems based upon measuring the amplitude of the return signals are prone to drift, the systems that depend upon measurement of "time" as the readout, such as the present temperature-gauge concept, are more appropriate.
- The present liquid-level and pressure sensors exhibit large insertion loss, thereby imposing need for rather complex electronics.

Subsequent study of this project has revealed several possible improvements capable of further enhancement of fiber-optic instrumentation feasibility.

## 1.5 RECOMMENDATIONS

The demonstration model reflects careful consideration of evaluated concepts, and associated electronics have been built using low-noise and noise-discrimination techniques. However, the additional improvements discussed in Sections 1.4.1 through 1.4.3 are also recommended:

## 1.5.1 Liquid-Level Sensor

Although much more efficient than previous ones, the newly designed sensors can be further modified by incorporation of improved mounting methods also developed by TRW (Ref. 2).

## 1.5.2 Pressure Gauge

The housing and the diaphram design should be modified to minimize the effect of temperature shock. Also, the design can be modified to incorporate a reference system to cancel drift effects. The reference is identical to the measuring system except that it measures the displacement of the diaphram near its circumference where pressure-induced diaphragm displacement is minimal. This would be a differential system in which the extraneous drifts are cancelled.

Future modification may incorporate a technique called the "Geode" system, or "Sign Around." These are essentially distance-measuring systems capable of measuring displacements as small as one micron. Because these are based on time-of-flight measurements, they are accurate and are devoid of extraneous drifts.

# 1.5.3 Temperature Sensor

The basic design of the temperature gauge is based upon the intrinsic transmissive properties of commonly available crystals. The readout of the present system measures the time taken by the power pulse to elevate the laser-diode temperature when the emission wavelength of the laser diode matches that of the sensing crystal. However, a better technique would be to measure the temperature of the junction of the emitting laser diode at the instant when wavelength matching takes place. The technique used for measuring instantaneous junction temperature of the laser diode is based upon the dependence of band-gap voltage of the forward-biased laser diode at a specific current, say 10 mA, or of one certainly less than the threshold current of the laser diode (Refs. 1-5).

#### 2. SENSOR SELECTION

Many sensor concepts were considered briefly, but almost immediately rejected because they were judged to be inappropriate for use in the cryogenic tank environment. An example is the concept of sensing liquid level via a float that is somehow connected to an optical readout system. Although it should be possible to devise a float system suitable for dealing with this problem, the mechanical complications appear to make this concept impractical and not worthy of serious consideration.

Because an exhaustive list of such "automatically" rejected concepts would not add to the clarity of this report, no such summary has been 'developed because the rejected concepts were not considered in any depth. Rather, the practical sensor concepts that were evolved from the early part of the Concept Study were evaluated in detail, as discussed in Sections 2.1 through 2.4.

#### 2.1 SPECIFIC CONCEPTS CONSIDERED

#### 2.1.1 Liquid-Level Sensor

Table 2-1 identifies the liquid-level sensor concepts considered, together with brief descriptions, the total evaluation score, and some pertinent comments. Table 2-2 lists details of the individual evaluations of each concept. The bare-fiber concept is clearly the performer choice, achieving the only perfect score of all the sensors. The main virtues of the bare-fiber level sensor are its extreme simplicity and its nondependence on absolute light levels due to the fact that the detector signal will either be on (when the fiber is not immersed) or off (when the fiber is immersed). Figure 2-1 is a sketch of the principle of the bare-fiber level sensor. The wetting response time is less than the time required to cover the fiber. The unwetting response time is slightly longer because of film effects, but laboratory tests on a model of this sensor have shown this effect to be minimal.

Table 2-1. Liquid-Level Sensor Concepts Considered

Concept Description	Evaluation Score*	Remarks
BARE FIBER (Light refracted from fiber when in in contact with liquid)	180	Extremely simple; residual film no problem
PRISM (Light ray reflected back to return fiber until liquid reaches prism and refracts beam into tank)	155	Good concept; residual film wetting would be a problem
NOTCHED CLADDING (Part of light refracted from notch when liquid is at notch level)	0	Dirt and residual droplets cause erratic behavior, causing elimination from further consideration

<sup>\*</sup>Maximum Score Possible = 180

Table 2-2. Evaluations of Point Level-Sensor Concepts

		Evalu	ation of Po Sensors	
Requirement/Goal	Scoring Basis	Bare Fiber	Prism	Notched Cladding
MANDATORY REQUIREMENTS				
Electrically Passive at Sensor	0, 10 (Yes, No)	10	10	10
Operates at Cryogenic Temperatures	0-10 (Judgment)	10	10	10
Can Perform Present Sensor Functions	0, 10	10	10	10
Can Operate Immersed or Nonimmersed	0, 10	10	5?	0**
Mechanically Simple and Rugged	0-10	10	5	
PERFORMANCE REQUIREMENTS				
Operating Range OK?	0-10	10	10	•
Sensitivity/Resolution OK?	0-10	10	5	
Response Time OK?	0-10	10	5	
Insensitive To				
Connector Reconnects?	0-10	10	10	
Source Intensity Variations?	0-10	10	10	
ΔT for Press Sensor?	0-10			
ΔP for Temp Sensor?	0-10			
Stability (over 200 cycles)?	0-10	10	10	
Remote Indication (100 ft) OK?	0-10	10	10	
ADDITIONAL CONSIDERATIONS				
Cost	0-10	10	5	
Weight	0-10	10	10	
Size	0-10	10	10	
Power Consumption	0-10	10	10	
Simple Data Processing and Readout	0-10	10	1C	
Interchangeability	0-10	10	10	
TOTAL SCORE*		180	155	0

<sup>\*</sup>Maximum Score Possible = 180

<sup>\*\*</sup>Automatic Elimination.

CONCEPT: With fiber cladding totally removed, external liquid "shorts out" transmitted light, thus indicating presence of liquid at fiber level.

REMARKS: Concept is point sensor, (i.e., On-Off indicator). Extremely simple sensor and readout.

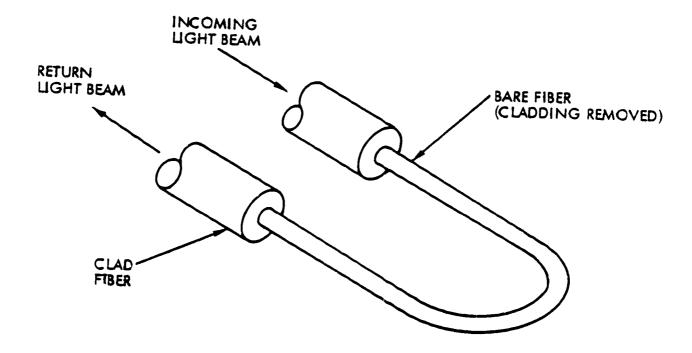


Figure 2-1. Recommended Liquid-Level Point Sensor Concept

## 2.1.2 Temperature Sensor

Table 2-3 shows the temperature-sensor concepts considered and Table 2-4 the details of the individual evaluations. The semiconductor absorption bandedge-shift concept is the most attractive sensor concept primarily because of its simplicity, which results from the fact that the sensor is a small chip inserted between the input and output fibers. Figure 2-2 illustrates the principles of the recommended temperature-sensor concept, and Figure 2-3 shows the temperature dependence of the absorption bandedge location for GaAs, given here as an example of the principle utilized.

Possible instrumentation concepts to implement the absorption bandedge concept for temperature measurement are shown in Figures 2-4(a) through 2-4(c), which correspond to the concepts listed under the semiconductor bandedge shifts addressed in Tables 2-3 and 2-4. All three instrument concepts were considered in detail, with the concept illustrated in Figure 2-4(a) the one chosen for the final instrumentation package.

## 2.1.3 Pressure Sensor

Pressure-sensor concepts considered are noted in Table 2-5 and details of the evaluations in Table 2-6. Because of their basic mechanical simplicity the concepts utilizing optical indication of the position of a pressure-sensitive diaphragm or bellows are preferred. Although, as indicated in Table 2-6, the digital encoder instrumentation concept was favored originally, a simpler system which senses the variations in light reflected from a pressure-deformable bundle ultimately was implemented. Variations in the output of the source light are sensed and normalized in accordance with obtained readings.

Table 2-3. Temperature-Sensor Concepts Considered

Concept Description	Evaluation Score*	Remarks
SEMICONDUCTOR BAND-EDGE SHIFT:		
Wavelength of Absorption Cut-Off is Temperature Dependent		
Match with Cut-Off of Reference		
Device;	176	Mechanically Simple;
Spectral Analysis of Output;	172	Simple But Higher Cost;
Ratio of Outputs From 2 Led's.	172	Requires 2 Sources.
PHOSPHORESCENCE:		
Time Decay of Light Pulse.	145	Inefficient?; Phosphores- cence Into $2\pi$ Ster.
VIBRONIC ABSORPTION:		
Ratio of Absorption In 2 Sidebands (Phonon Creation/Destruction).	144	Little Known Interaction; Good Potential, Requires More Research.
TWO-COLOR INTERFERENCE COUPLER:		
Spacing Varies With Expansion; Ratio of 2-Color Signals Temperature Dependent.	138	Difficult To Match Fiber N.A., Surface Finish, etc.
FABRY-PEROT:		
Spectral Analysis of Transmitted Intensity.	136	Requires High Spectral Resolution, Therefore High Cost.
THERMOSTAT BIMETAL:		•
Optical Readout of Motion	110	Expansion Coefficient Low at Cryogenic Temperatures; Many Fibers In Liquid.

<sup>\*</sup>Maximum Score Possible = 190

Table 2-4. Evaluation of Temperature-Sensor Concepts

		Semicond	Semiconductor Bandedoe Shift	Shift					
		Match Cut-off of	Spectral	Ratio Outputs			Two-Color		
	Scoring	Reference	Analysis	from	Phosphor-	Vibronics	Interference	Fabry.	Thermostat
Requirement/Goal	Basis	Device	of Output	2 LED's	escence	Absorption	Coupler	Perot	Bimetal
MANDATORY REQUIREMENTS								:	
Electrically Passive at Sensor	0, 10 (Yes No)	10	01	0	0	01	10	0	10
Operates at Crystophin Temporary	01.0	ç	ç	Ç	ı	ı	(		ı
Operates at Cryogenic Temperatures	0-10 (Judgment)	2	5	0	ഹ	လ	œ	ഗ	ro.
Can Perform Present Sensor Reactions	0, 10	5	5	0	ហ	ß	S	5	ß
Can Operate Immersed or Nonimmersed	0, 10	5	5	10	0	0	5	01	10
Mechanically Simple and Rugged	0-10	9	9	œ	0	01	2	ഗ	7
PERFORMANCE REQUIREMENTS									
Operating Range OK?	0-10	ω	80	ω	55	25	œ	œ	ß
Sensitivity/Resolution OK?	0-10	7	7	7	25	55	œ	5	7
Response Time OK?	0.10	6	10	တ	œ	œ	∞	œ	w
Insensitive to									
Connector Reconnects?	0.10	5	5	<b>∞</b>	5	œ	ဌာ	ß	ß
Source Intensity Variations?	0-10	10	01	œ	0,	മ	7	7	S
∆T for Press Sensor?	0-10	ı	1	ı	ı	ı	ı	ı	
∆P for Temp Sensor?	0-10	01	01	5	0	01	6	თ	10
Stability (Over 200 Cycles)?	0-10	10	01	0	25	01	S	œ	S
Remote Dedication (100 Ft) OK?	0-10	0	5	5	2	လ	ω	æ	S
ADDITIONAL CONSIDERATIONS									
Cost	0-10	S	ഹ	œ	œ	ĸ	ഹ	ß	ß
Weight	0-10	0	ω	0	6	တ	6	88	လ
Size	0-10	5	5	5	œ	o,	8	7	S
Power Consumption	0-10	ω	80	œ	တ	œ	80	വ	ro
Simple Data Processing and Readout	0-10	œ	0	<b>co</b>	5	7	2	വ	o
Interchangeability	0-10	5	5	5	œ	10	10	ω	2
TOTAL SCORE		: 76	176	172	145	144	138	136	110

Maximum Score Possible ≈ 190

CONCEPT: Light absorbed by semiconductors or insulators dependent on intrinsic or impurity absorption; absorption bandedge a sharp cut-off and temperature-dependent.

REMARKS: Simple, small-size sensor; various instrument concepts available

EXAMPLE: Transmission and absorption of emission spectrum of HP HEMT-6000 LED at 25°C shown with selenium crystal, also at 25°C.

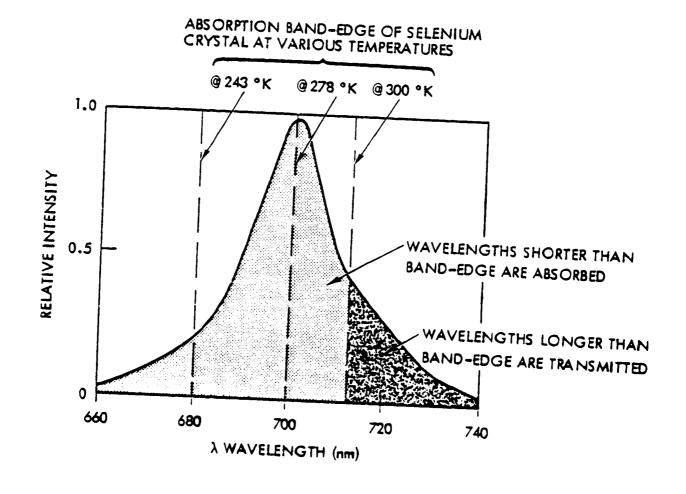


Figure 2-2. Recommended Temperature-Sensor Concept

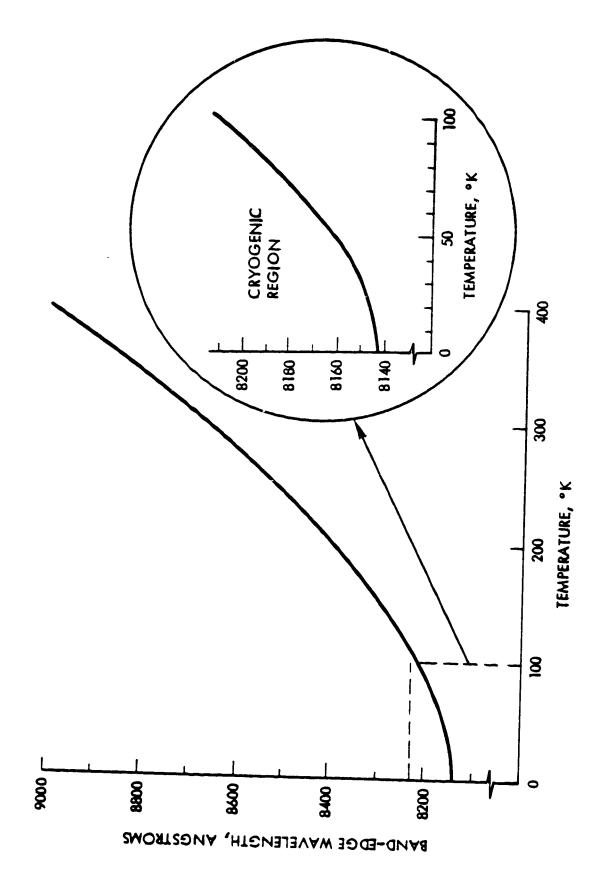


Figure 2-3. Temperature Dependence of Absorption Bandedge Location for GaAs

INSTRUMENT CONCEPT: LI

LIGHT BEAM FROM LED IS PASSED THROUGH BOTH REFERENCE SEMICONDUCTOR (AT CALIBRATED, CYCLED TEMPERATURE  $_{\rm R}^{\rm R}$ ) AND SENSOR (AT  $_{\rm S}^{\rm S}$ ). WHEN  $_{\rm I}^{\rm R}$  <  $_{\rm I}^{\rm S}$  ABSORPTION CONTROLLED BY SENSOR; WHEN  $_{\rm I}^{\rm R}$  >  $_{\rm I}^{\rm S}$  ABSORPTION CONTROLLED BY REFERENCE. KNEE OF INTENSITY-TIME RESPONSE DETERMINES  $_{\rm I}^{\rm S}$  =  $_{\rm I}^{\rm R}$ .

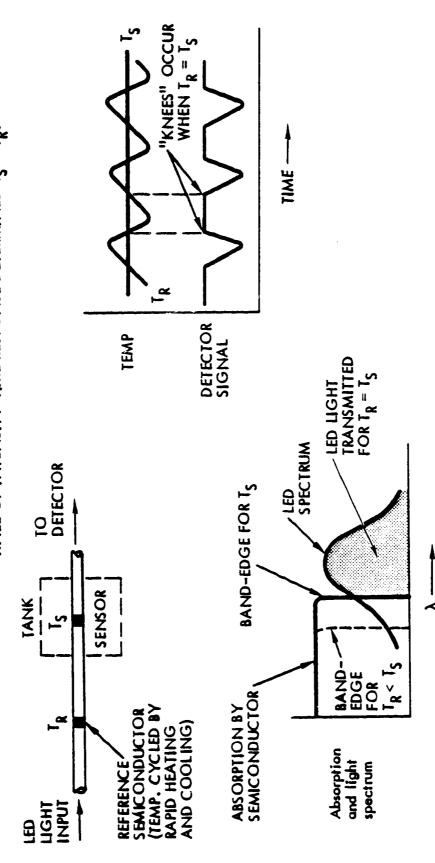


Figure 2.4(a). Instrument Concept for Semiconductor Bandedge Temperature Sensor: Reference Temperature Cycling

TRANSMISSION CUT-OFF INDICATES SENSOR TEMPERATURE, 15. TRANSMISSION OF LIGHT BEAM FROM LED IS CONTROLLED BY ABSORPTION BAND-EDGE OF SENSOR, WAVELENGTH OF INSTRUMENT CONCEPT.

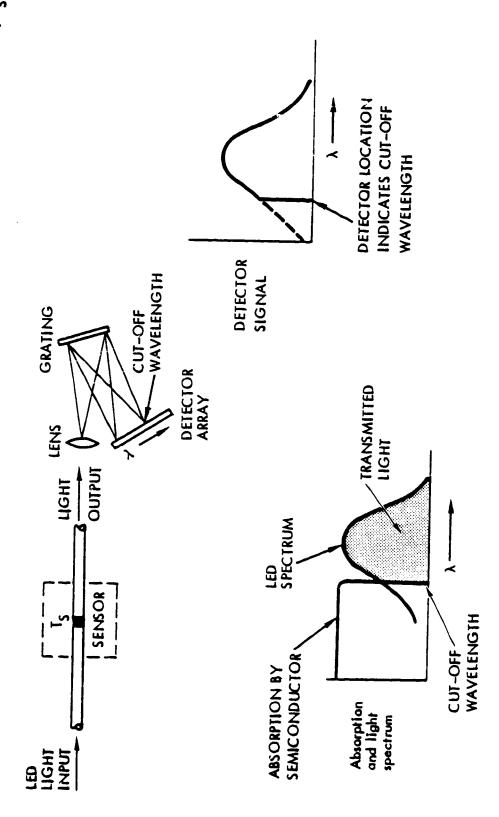


Figure 2-4(b). Instrument Concept for Semiconductor Bandedge Temperature Sensor: Spectral Analysis

RATIO OF TRANSMITTED LIGHT FROM LED'S INDICATED SENSOR TEMPERATURE INSTRUMENT CONCEPT, TRANSMISSION OF ALTERNATE BEAMS FROM 2 DIFFERENT-COLOR LED'S IS CONTROLLED BY ABSORPTION BAND-EDGE OF SEN! OR,

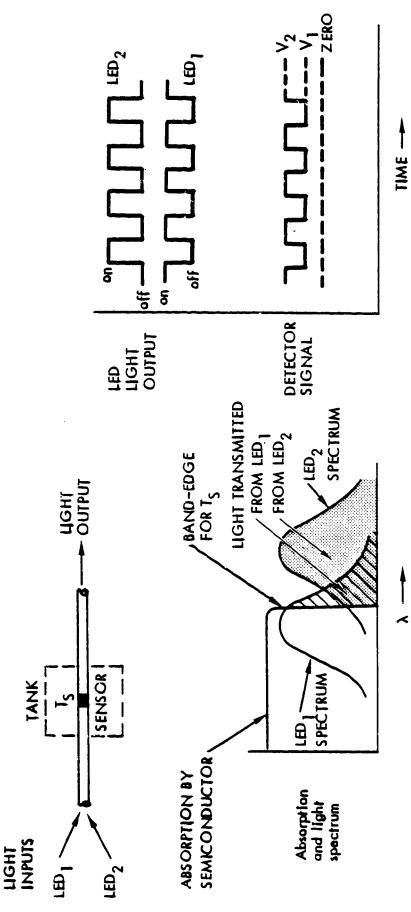


Figure 2-4(c). Instrument Concept for Semiconductor Bandedge Temperature Sensor: Ratio of 2-Color Outputs

Table 2-5. Pressure-Sensor Concepts Considered

Concept Description	Evaluation Score*	Remarks
MOTION OF DIAPHRAGM OR BELLOWS		All Motion Devices Simple and Rugged
Optical Indication of Position		
Digital Encoder	166	No Ratioing Needed; Sensitive
Fiber Ribbon	164	Requires Many Fibers
Variable Size Holes	162	Requires Intensity Ratioing
Linear Taper Absorber	157	Requires Intensity Ratioing
Linear Density Filter	156	Requires Intensity Ratioing
Moire Grid	151	Zero Point Uncertain
Fiber Bundle and Diaphragm	147	Requires Ratioing, Initial Calibration
PRESSURE ACTIVATED SPECTROMETER		
Location of Dispersed Beam Varies with Pressure	134	Spectral Variation of Fiber Ribbon Indicator; Temperature Sensitive
PHOTOELASTIC METHOD		
Variable Transmission Through Stressed Birefringent Crystals	93	Mechanically Complex; Temperature Sensitive

<sup>\*</sup>Maximum Score Possible = 180

Table 2-6. Evaluation of Pressure Sensor Concepts

							Action	Action of Diaphragm or Bellows	or Bellows	
									Pressure	
						Linear		Fiber-	Activated	Photo-
	Scoring	Digital	Fiber	Hole	Linear	Density	Moire	Bundle	Spectro-	Elastic
Requirement/Goal	Basis	Encoder	Ribbon	Indicator	Taper	Filter	Grid	Diaphragm	meter	Method
MANDATORY REQUIREMENTS										
Electrically Passive at Sensor	0, 10	10	0	5	6	10	10	10	0	0
	(Yes, No)									
Operates at Cryogenic Temperatures	0.10	5	5	10	10	5	0	10	5	25
	(Judgment)									
Can Perform Present Sensor Functions	0, 10	5	0	0	0	0	10	10	0	2
Can Operate Immersed or Nonimmersed	0, 10	ł	1	ı	ı	ł	I	ı	İ	YES**
Mechanically Simple and Rugged	0-10	6	10	0	2	ß	ß	ĸ	ιΩ	2
PERFORMANCE REQUIREMENTS								•	•	]
Operating Range OK?	0-10	10	9	10	5	10	0	10	10	LC.
Sensitivity/Resolution OK?	0-10	ω	80	7	∞	œ	5	00	, LC	LC.
Response Time OK?	0-10	01	10	10	0	0	5	10	6	2
Insensitive to									•	)
Connector Reconnects?	0-10	5	0	5	œ	œ	9	7	œ	
Source Integrity Variations?	0-10	10	0	01	œ	œ	∞	7	• •	. LC
∆T for Press Sensor?	0-10	<b>∞</b>	ø	œ	œ	œ	ഹ	- α	2 0	0 0
ΔP for Temp Sensor?	0-10	ı	ı	ı	1	' I	' I	' I	۱ ۱	<del></del>
Stability (over 200 cycles)?		5	5	5	5	01	10	∞	œ	-LC
Remote Indication (100 ft) OK?	0.10	01	5	10	5	5	5	5	0,	0
ADDITIONAL CONSIDERATIONS										
Cost	0-10	œ	တ	80	œ	œ	တ	7	7	ıc
Weight	0-10	œ	œ	<b>∞</b>	∞	œ	7	9	വ	2
Size	0-10	O	7	6	6	6	7	S	2	2
Power Consumption	0-10	œ	6	7	7	7	æ	7	7	
Simple Data Processing and Readout	0-10	<b>6</b> 0	ഹ	2	œ	7	മ	9	. ц	m
Interchangeability	0-10	01	0	0	0	5	0	10	10	S
TOTAL SCORE*		166	<del>2</del>	162	157	156	151	147	134	93

<sup>\*</sup>Maximum Score Possible = 180
\*\*On!y photoelastic method requires immersed sensor

#### 2.2 OVERVIEW OF STUDY RESULTS

After an initial period wherein essentially all types of sensor concepts were considered, the range of concepts was narrowed to those identified in Figure 2-5. Final recommendations were based on the numerical scoring procedure indicated in Table 2-7, the evaluations being weighted strongly in favor of practical systems which operate in a cryogenic environment. The selected concepts have been incorporated in the demonstration model to permit further evaluation.

#### 2.3 CONCEPTS SELECTED

This section identifies sensors selected for final development from the various sensor concepts previously considered in Section 2.1. Specific test data that have been accumulated on the developed sensors are summarized in Section 4.

## 2.3.1 Liquid-Level Sensor

The selected fluid-level sensor concept is based upon measuring optical attenuation through a glass rod used as the level sensor, as shown in Figure 2-6. Attenuation through the sensor is a function of the refractive index of the material surrounding the glars rod. Because liquids generally have higher refractive index, attenuation through the sensor increases when the sensor is submerged in the liquid. This concept in its simplest form works very well for liquids having high refractive indices, such as water, gasoline, oil, etc. However, liquid-level sensors for low refractive indices such as those of LN<sub>2</sub> or LH<sub>2</sub> must be designed with several concepts given consideration. The anatomy of the present sensor design is elaborated below.

Figure 2-7 is a schematic of a system in which light is coupled from an input fiber to an output fiber through a straight unclad coupling rod. In the following discussion it is assumed that the index of refraction of the fiber-optic core is the same as that of the rod, that there are no reflection losses, and that  $n_3 < n_2$ . The latter is true for all fibers and fuels of interest to this program.

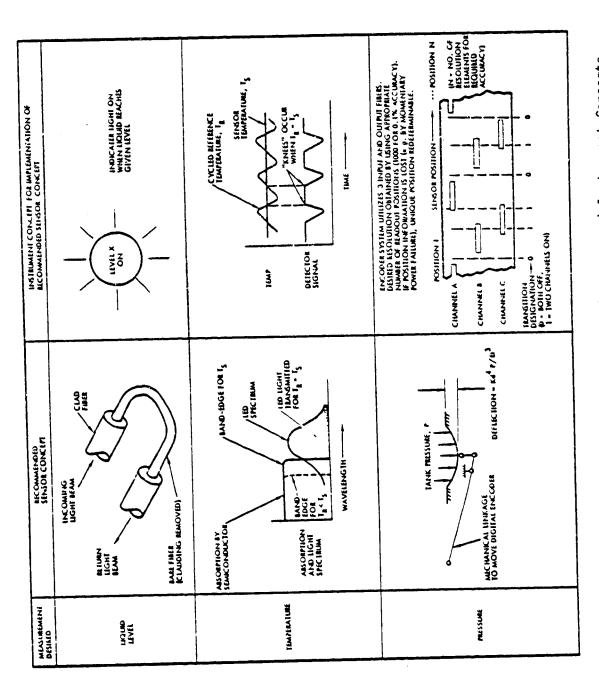


Figure 2-5. Summary of Recommended Sensor and Instrument Concepts

Table 2-7. Sensor Scoring Evaluation

Requirement/Goal	Scoring Basis*
MANDATORY REQUIREMENTS Electrically Passive at Sensor	0, 10 (Yes, No)
Operates at Cryogenic Temperatures Can Perform Present Sensor Functions Can Operate Immersed or Non-Immersed Mechanically Simple and Rugged	0-10 (Judgement) 0, 10 0, 10 0-10
PERFORMANCE REQUIREMENTS Operating Range O.K.? Sensitivity/Resolution O.K.? Response Time O.K.? Insensitive To	0-10 0-10 0-10
Connector Reconnects?  Source Intensity Variations? $\Delta T$ For Press Sensor? $\Delta P$ For Temp Sensor?  Stability (Over 200 Cycles)?  Remote Indication (100 Ft) O.K.?	0-10 0-10 0-10 0-10 0-10 0-10
ADDITIONAL CONSIDERATIONS Cost Weight Size Power Consumption Simple Data Processing and Readout Interchangeability	0-10 0-10 0-10 0-10 0-10 0-10

<sup>\*</sup>Maximum Score Possible = 200

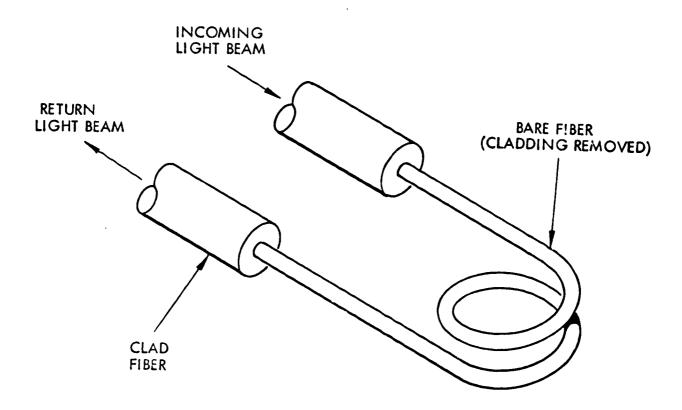


Figure 2-6. Selected Fluid-Level Sensor Concept

The light entering the rod will have a numerical aperture given by

$$\sin\theta_1 = (n_1^2 \cdot n_2^2)^{1/2}$$

where  $\theta_1$  is the angle between the extreme ray propagated by the fiber and the fiber axis (Ref. 1). Since N<sub>3</sub>, the refractive index of the medium surrounding the rod, is less than n<sub>2</sub> for all cryogenic fuels, the numerical aperture of the rod, or

$$\sin\theta_2 = (n_1^2 - n_3^2)^{1/2} > \sin\theta_1$$

is greater than that of the fiber. Therefore if the rod is aligned to the fiber it will transmit all of the incoming light without loss whether or not it is surrounded by the liquid. If the diameters of the fibers and

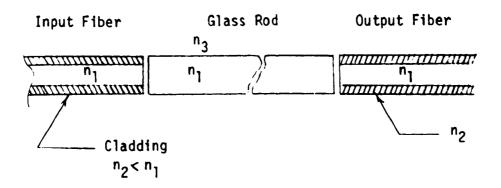


Figure 2-7. Schematic of Light Coupling Between Input and Output Fibers

the rod are not matched there will be coupling losses associated with the system, but these losses will be independent of the medium surrounding the sensor rod. Consequently, a straight in-line sensor as shown in Figure 2-7 cannot be used as a liquid-level gauge.

It also would be possible to introduce a scattering center within the rod to induce higher-order modes, whereby the fraction of input light that is transmitted by the rod would be a function of the index of refraction of the medium because the higher-order modes created by the scattering center would be cast through the walls of the rod. However, when  $n_3 < n_2$ , as is always the case here, the final solution of the amount of light transmitted to the receiver would be governed by the numerical aperture of the output fiber (which is smaller than that of the rod), and the signal received would again be independent of the medium surrounding the sensor rod. Therefore this scheme also cannot be used to detect the presence or the level of a fluid.

A multiplicity of scattering centers of course would serve to couple energy between modes, so that the amount of light accepted and transmitted by the output fiber would be a function of the index of refraction of the medium surrounding the rod. However, in practice it is difficult to fabricate rods with the proper number and distribution of scattering centers, and TRW experience indicates that commercial glass, poor as it is, does not have a sufficient number.

It thus is necessary to produce in a controlled manner effective means of inducing higher-order modes as well as of coupling energy between modes. The simplest technique for accomplishing this is to form a number of bends in the rod, each bend serving to mix incoming modes, to outcouple high-order modes, and to induce new modes. The amount of light exiting from the rod within the numerical aperture of the return fiber then will be a function of the index of refraction of the medium surrounding the rod.

The optimum number of bends and the radii of curvature of the bends must be determined experimentally. The closer the index of refraction of the liquid is to that of air, the greater are the number of bends

required to accentuate the difference between air and the liquid. TRW has found through a number of experiments on a variety of level-sensor configurations that the configuration shown in Figure 2-6 is optimal for the present system.

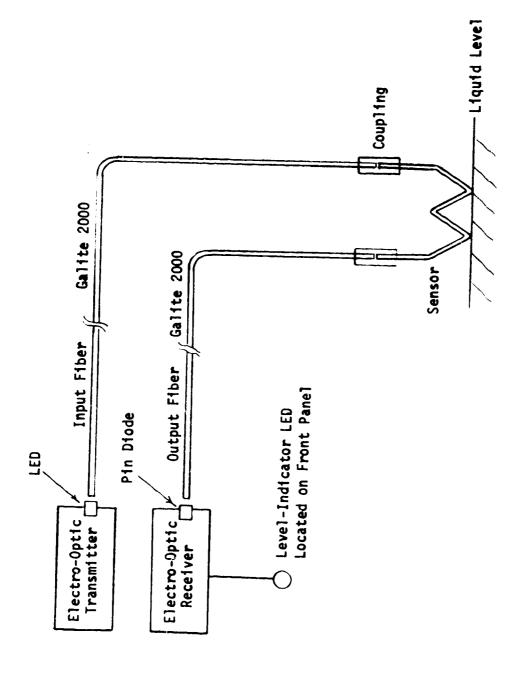
Insertion loss of this sensor in air is 18 db. When immersed in liquid nitrogen the signal loss is 28 dB, yielding a 10-dB change in the signal in going from no-fluid to fluid environment. The signal loss in liquid hydrogen is estimated to be 22 dB, or a 4-dB change in signal due to the presence of the fluid. Such change magnitudes are readily detected and indicated by the recommended system. Figure 2-8 is a schematic of the entire level-sensing system.

#### 2.3.2 Pressure Sensor

Figure 2-9 is a sketch of the recommended pressure-sensor configuration. The fiber-optic cable has been bifurcated so that some (about one-half) of the fibers are used for input and the remainder for output. Light is coupled from the input fibers to the output fibers by reflection off the thin-end plate diaphragm. The curvature of this diaphragm is determined by the pressure external to the sensor (i.e., flat at atmospheric pressure and concave inward at higher pressures). This change in shape of the reflecting surface with pressure changes the coupling between input and output fibers, thus producing a signal change as a function of pressure. Locations of the fiber end faces relative to the diaphragm are set and locked during assembly and test. The entire sensor is made of Invar to minimize thermal effects. Figure 2-10 is a schematic of the electronic circuitry associated with this sensor.

## 2.3.3 Temperature Sensor

This sensor is the most sophisticated of the three in that its operation is based upon the fact that the position of the optical absorption (or transmission) edge of the semiconductor crystal is a well-defined function of temperature. A schematic of this phenomenon is shown in Figure 2-11, wherein T is the temperature of the LED and t is the



j

Figure 2-8. Schematic of TRW Level-Sensing System

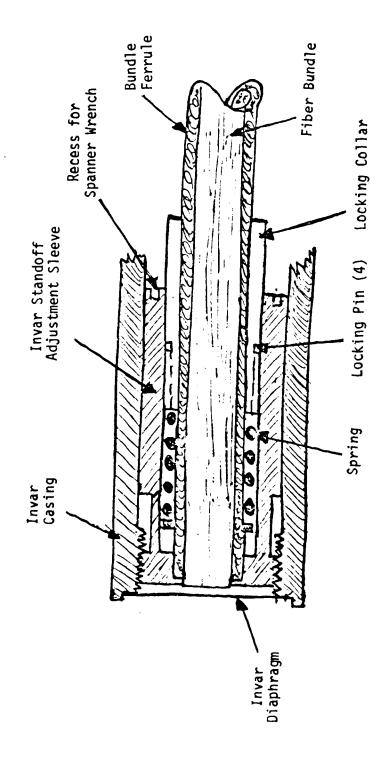


Figure 2-9. Recommended Pressure-Sensor Configuration

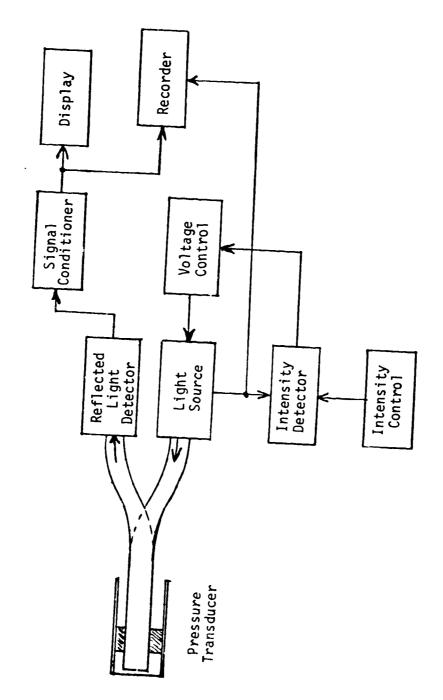


Figure 2-10. Electronic Circuit Associated with Pressure Sensor

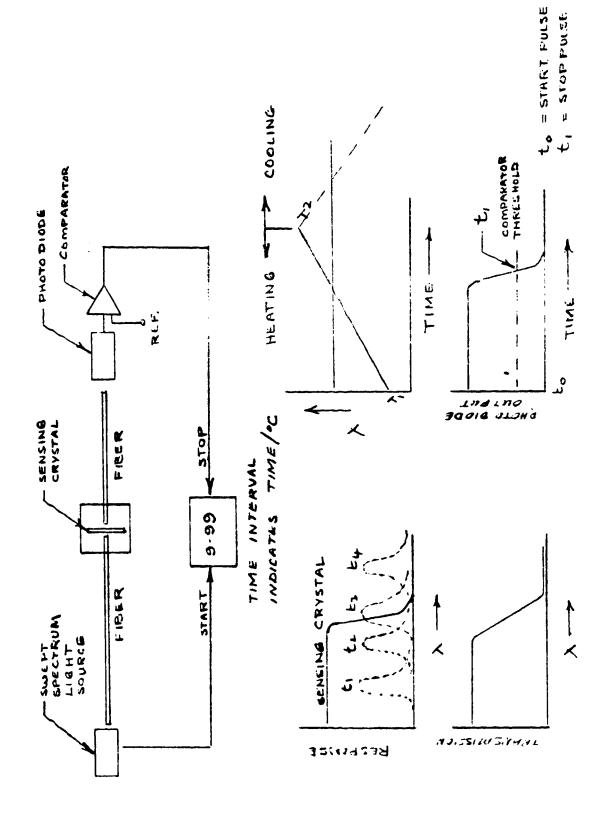


Figure 2-11. Temperature-Tuned LED Output Sweeping Through Absorption Edge of Sensor Crystal

length of time after the sensor crystal has turned on and begins to heat up due to the electrical energy dissipated within it. Thus, if an appropriate crystal is immersed in the fluid of interest and the wavelength of the absorption edge determined, the temperature of the crystal (and hence of the fluid) is known.

A number of techniques for locating the position of the absorption edge use an optical source to sweep through the wavelengths of interest. TRW chose to measure the optical transmission of the sensor crystal using a narrow-band tunable optical source. The wavelength at which there is a rapid change in the transmission is the cutoff wavelength which is directly analogous to the crystal temperature.

Gallium arsenide (GaAs) laser diodes satisfy applicable requirements, and they may be operated at room temperature. Because room-temperature GaAs lasers emit around 0.87 microns (exact wavelength depends upon doping) the sensor crystal must have an absorption edge near that wavelength when it is at cryogenic temperature. Indium phosphide (InP) is such a crystal, as is evident from comparing the GaAs data with the InP data of Figure 2-12. It should be noted that at 300°K the output of a GaAs laser is at a wavelength slightly shorter than the bandedge wavelength of InP at cryogenic temperatures, which makes it possible to tune the GaAs output through the InP absorption edge by heating the laser. Heating is highly preferred over cooling because heating can be accomplished by the laser drive current itself.

The simplest tunable optical approach is to use a pulsed laser diode driven in such a manner that during its ON pulse it heats up, causing its output wavelength to increase. Figure 2-13, which shows this process schematically, is a sketch of the complete temperature-sensing system and of associated input and output signals.

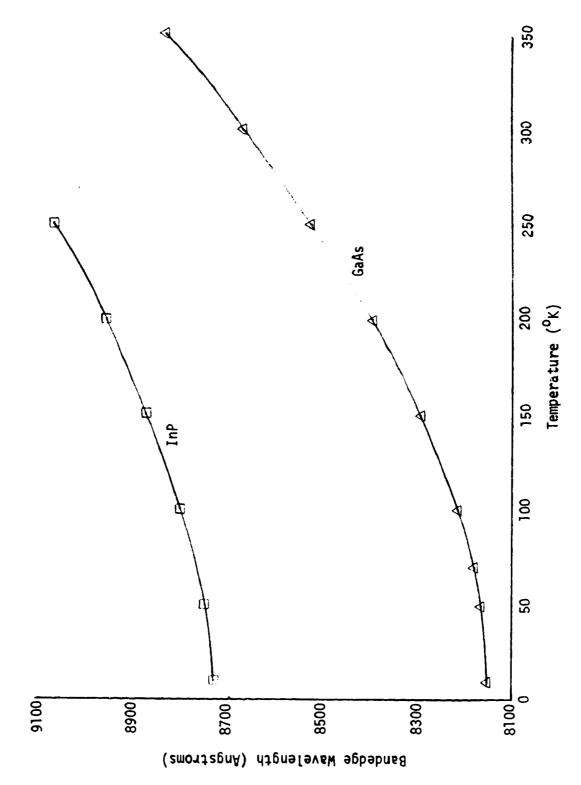


Figure 2-12. Data Bandedge of Indium Phosphide and Gallium Arsenide vs. Temperature

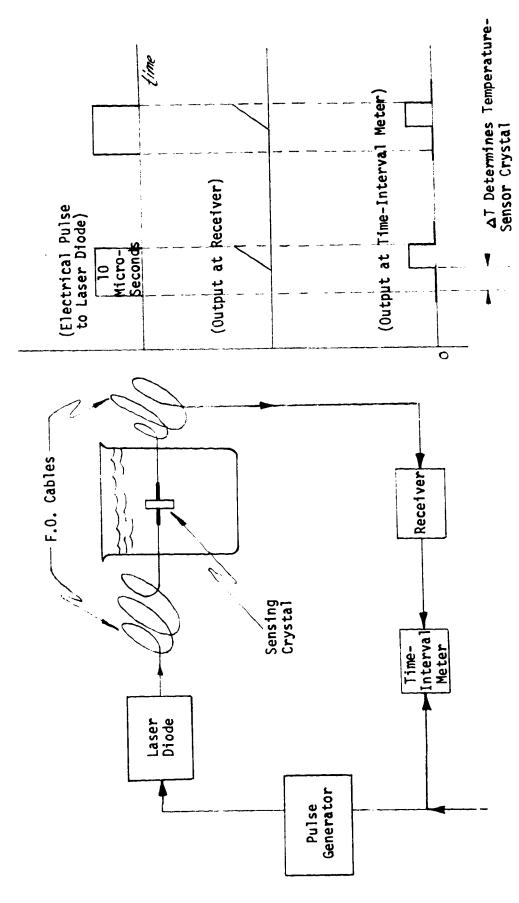


Figure 2-13. Schematic of Temperature-Sensing System and Associated Input/Output Waveforms

#### 3. EXPERIMENTAL MODEL

An experimental model was built to demonstrate and evaluate the performance of all three sensors. As shown in Figure 3-1, the model consists of a console cabinet with a cryogenic Dewar affixed to its side. The necessary electronics is behind the panels mounted in the sloped cut-out. The Dewar can be detached and operated as far away from the cabinet as 30 feet. The only connection between cabinet and Dewar is via fiber cables. The experimental model is self-contained, however, and therefore helium gas and liquid nitrogen must be applied during test. The electronics, self-contained with internal power supplies, can be operated by a 120V line. The total power required is less than 120 watts. Various test jacks located on the rear panel of the electronics module are connected to a standard oscilloscope to monitor applicable signals. To operate the demonstration model the following items are required:

- Helium gas cylinder with two pressure control valves
- Liquid nitrogen cylinder with stop valve
- Standard triggered oscilloscope

#### 3.1 PERFORMANCE REQUIREMENTS

Sensor performance requirements were dictated principally by the test procedures specified in a previously issued TRW document identifying test plans and procedures for fiber-optic instrumentation (Ref. 3).

In accordance with Section 2.1 of the above document, the three sensors are mounted on a sensor chamber attached to a 2-ft section of galvanized tubing (sting assembly) as shown in Figure 3-2. This tube can be lowered into and out of the Dewar to allow three sensors to be inserted into a quiescent bath of  $LN_2$  contained in the cryogenic Dewar. The Dewar is designed to withstand pressures of 40 psig, which is higher than the required test pressure of 15 psig. For safety purposes, a 15-psig limit valve has been incorporated.

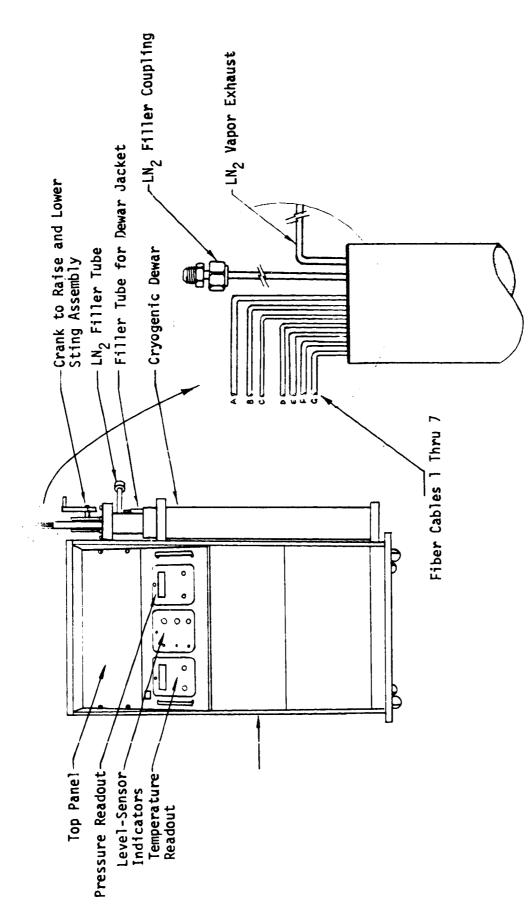


Figure 3-1. Experimental Model

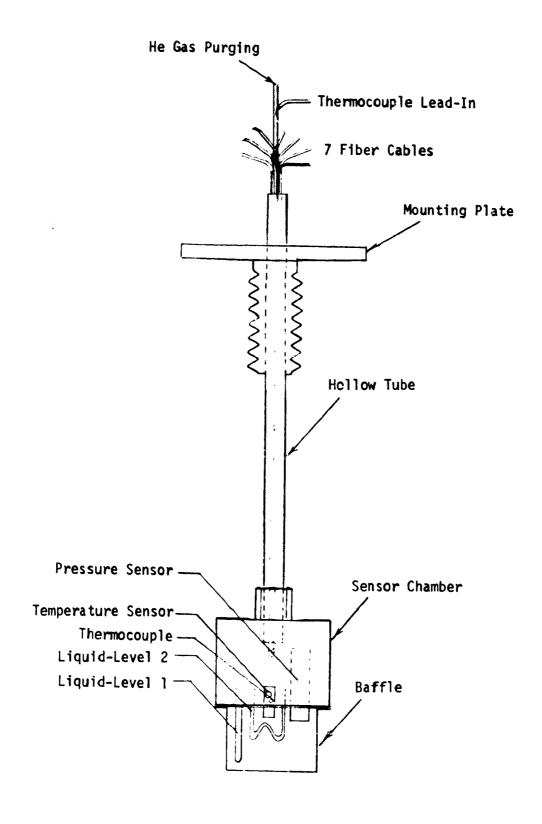


Figure 3-2. Sting Assembly

Sensor performance goals versus achieved performance of the experimental model are noted in following Sections 3.1.1 through 3.1.11.

# 3.1.1 Environment

The cryogenic Dewar of the experimental model allows testing of the sensors at 77°K.

# 3.1.2 Operation Mode

The sensor chamber and the sting assembly can be lowered or raised five inches. This provides sufficient height variation to cover and uncover the sensors during dry-to-wet or wet-to dry operation.

# 3.1.3 Level Accuracy

As discussed under level-sensor design methodology, the decision height-of-level indication is less than the diameter of the glass rod used to construct the liquid sensor. In the experimental model the sensors are of 1-mm diameter rod. Therefore the level accuracy is 1 mm or less. The diameter of the smallest tank that can provide 0.1 percent level accuracy by volume would be 10 cm provided the tank is cylindrical.

# 3.1.4 Temperature Range

The experimental model operates at 77  $^{\circ}$ K. Its design permits operation down to 20  $^{\circ}$ K, although no tests have been performed to evaluate operation below 77  $^{\circ}$ K.

# 3.1.5 Pressure Range

Although the Dewar is designed for 350 psig, no tests have been performed with pressures higher than 45 psig.

# 3.1.6 <u>Time Response</u>

Level Sensor: "On" time is limited only by the time constants of the receive electronics, whereas "Off" time is dependent upon the time needed for the sensors to dry off. Care has been exercised in designing the shape of the sensor to preclude clinging of liquid. Assessment of success in meeting this requirement was accomplished by monitoring the output signal of the sensor on the scope while the Dewar was being jolted. Best estimate is the response time was approximately 150 msec "On" time and 300 msec "Off" time.

- O Pressure Gage: The limiting factors for the pressure gage time response are the mechanical time constants of the diaphragm. With appropriate analysis it can be shown that the time response is less than 100 msec.
- O Temperature Gage: In the present setup, the light source is a pulsed laser diode that is pulsed once every 750 msec (i.e., one reading every 750 msec). The response of the temperature gauge gauge therefore is 750 msec. This performance can be improved by selecting different laser diodes that would operate at pulsing frequency of greater than 10 Hz.

### 3.1.7 Sensor Installation Accuracy

All sensors are designed to be affixed by cyogenic epoxy onto the reference platform. A simple jig can be used to assure installation within an accuracy of 0.1 inch.

### 3.1.8 Sensor Location

The only factor limiting the distance between the location of the sensor and the electronics is the lineal attenuation of the fiber. The fiber bundle being used presently is Galite 2000, which has an attenuation of 650 dB/km. The worst-case power margin of only 15 dB of the level sensors includes the loss of 60 feet of fiber. Extra length of fiber will increase the system loss by 3 dB, which would leave 12 dB as an adequate power margion.

In the experimental model, one of the level sensors is fitted with 30 meters of fiber each way, which clearly demonstrates the capability of sensors operating at a distance of 100 feet.

### 3.1.9 Flow Location

This performance goal relates to mechanical strength of the various sensors. Mountings of the sensors are sufficiently sturdy to withstand 27 fps liquid flow. Sensors are further projected by a baffle that is installed around the sensors.

#### 3.1.10 Sensor Lifetime

- o Sensor lifetime is principally dependent upon the following moving mechanical parts:
  - Bellows on the sting.
  - Rack and pinion installed on the sting to raise and lower the sensors in the Dewar.
  - Crank handle mounted on the Dewar.
  - Rubber gasket between the Dewar and the cover.
  - Tube joints connecting LN $_2$  and helium to the Dewar.
- o The steel bellows can fail due to any puncture in it. Therefore, unless it is punctured accidentially, it can last for 200 cycles and fatigue failure can be precluded for 200 cycles.
- o Rack and pinion can be damaged if the sting is moved forcibly while frozen due to frozen-water moisture in the sting guides. Correct procedure would be application of heat with a small heat gun until the sting frees itself. If accidentially damaged before 200 cycles, the rack and pinion can be repaired using very ordinary mechanical components.
- o Crank handle is sufficiently sturdy to last for 200 cycles. Rubber gasket should be replaced as necessary.
- o Tube joints should be replaced as necessary.

### 3.1.11 Lifetime of Electronics

Electronic components lifetime is estimated to be at least one year of continuous use. Degradation of LEDs and of the laser diode within one year have been considered in the design, and appropriate margins have been incorporated. However, periodic failures have not been precluded, and replacement parts are easily available.

#### 3.2 TEST PROCEDURES

Setup and measuring procedures are complicated only by the two requirements that the Dewar must be free of moisture and that the pressure gauge must be calibrated differently at different temperatures. It must be noted that presence of moisture in the Dewar interferes with operation of the liquid-level sensor because of moisture condensation.

# 3.2.1 <u>Initial Setup</u>

The demonstration model is broken down for shipment into two major elements consisting of the console and the sting assembly. For reassembly, the Dewar should first be mounted on the side of the cabinet, using the brackets and hardware that are provided. Proper care must be exercised in handling of the Dewar because the fiber cables emerging from the Dewar and from the sensor assembly within the Dewar are relatively fragile.

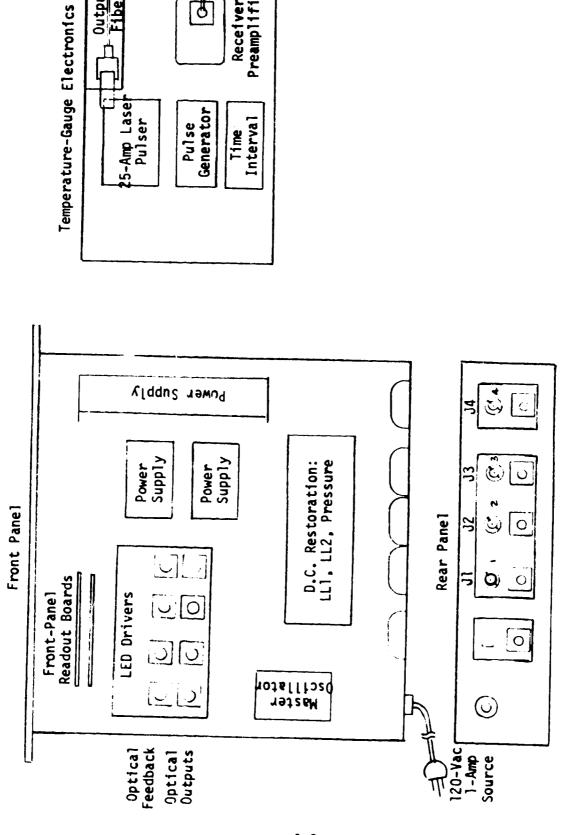
After the Dewar has been installed, the seven fiber cables together with the thermocouple lead-in can be passed through the hole on the side of the cabinet. The surplus lengths of the cables can then be coiled and stowed outside the cabinet in the manner indicated in Figure 3-3. All fiber connections should be installed as shown. Because the fiber-optic connectors are made of plastic and are therefore fragile, particular care must be taken to avoid misalignment of connections and resulting stripped threads. Proper installation of Dewar and of the fiber cables may require as much as four to five hours.

After all fiber connections have been made the ac power cord may be plugged into any 117-Vac outlet and the electronics turned on by actuating the power switch on the front panel. The two digital meters should then light up.

The final step is to purge and to cool down the Dewar.

# 3.2.2 Cool-Down and Moisture-Purging Procedure

- Inspect empty test Dewar interior for moisture droplets. Wipe Dewar walls and floor if any visible moisture is present.
- 2. Insert sensor module into test Dewar. Secure cap with screws provided.
- 3. Open test Dewar exhaust valve.
- 4. Purge test Dewar with gaseous helium via its fill tube for about 15 minutes.
- 5. Simultaneously purge sensor module interior with gaseous helium via the Optical Cable Access tube Reduce gas flow to a "trickle" after 15 minutes and continue the "trickle purge" throughout



Receiver Preamplifier

Output Fiber

Experimental-Model Fiber-Optic Connections Figure 3-3.

demonstration. When testing temperature sensor withdraw purge tube to within 3 inches of top of access tube to prevent spurious heating signal from helium flow.

- 6. Close test Dewar exhaust valve and built up 5 psig in test Dewar, and then shut off helium flow to fill tube.
- 7. Remove helium input line from fill tube and connect liquid nitrogen transfer tube.
- 8. Crack nut on transfer tube at supply Dewar, re-open test Dewar exhaust valve, and purge transfer tube by directing helium back through the test Dewar exhaust line. (For this operation the helium input line can be hand-held to exhaust port.) After purging re-tighten nut on transfer tube.
- 9. Position sensor module about midway on the gear rack positioning mechanism.
- 10. Open test Dewar exhaust valve.
- 11. Fill test Dewar "slowly" with liquid nitrogen to desired level, as indicated by level sensors.
- 12. Insert sensor module about 2-1/2 inches into liquid nitrogen and allow about <u>60</u> minutes to achieve thermal equilibrium. Top off liquid nitrogen as required during equilibration. It will be necessary to top off three or four times.
- 13. Proceed with all demonstrations. Refill as necessary.

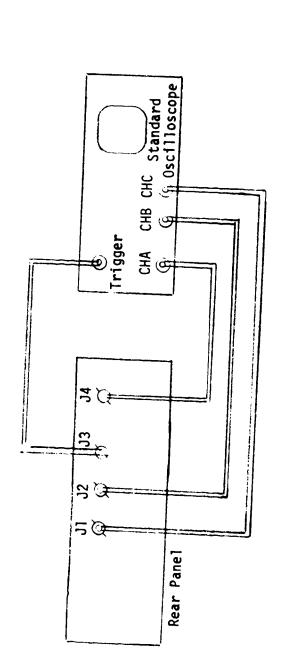
#### SPECIAL NOTES ON COOL-DOWN PROCEDURE

- Item 1: If  $LN_2$  is left in Dewar ice will form inside.
- Item 2: GN<sub>2</sub> is satisfactory for initial purge, but after liquid nitrogen is in Dewar any further purging should be done with GHe because of its lower freezing point.
- Item 6: Positive pressure inside Dewar keeps moisture out.
- Item 11: Filling takes approximately 9 minutes.

### 3.2.3 <u>Calibration Procedures</u>

Calibration procedures consist mainly of adjusting the output levels of the LEDs that provide optical power to the sensors. The criterion of proper adjustment is to ensure particular voltage level at the output of the receiver preamplifier, which in this case is 1-V peak. The liquidlevel sensors and the pressure-gauge detection circuits then are adjusted to operate properly with 1-V signals. Once these levels are set no future adjustments are necessary because all the LEDs incorporate an optical feedback arrangement which keeps LED output constant. However, readjustment will become necessary whenever the fiber optic connectors are remated. Calibration can be accomplished using a triggered oscilloscope. A 3-channel oscilloscope unit would be most appropriate in order to allow the two level sensors and the pressure sensor to be adjusted simultaneously. The temperature gauge does not require any calibration. The oscilloscope should be connected as shown in Figure 3-4. With the levels of each output set at a 1-V peak with the trimpots located on the front panel, the pressure linearity pot should be adjusted to cause the pressure meter to read zero with zero pressure. The temperature gauge requires no adjustment because it is dependent upon time measurements rather than on voltage measurements.

A thermocouple is attached to the fiber-optic temperature gauge for calibration purposes. The thermocouple leads should be connected to a dc millivolt meter. The output of the fiber-optic temperature gauge can also be monitored on an oscilloscope, but should be done on a different unit because the repetitition rate of the temperature gauge is approximately 1 pps whereas those of the other gauges are 1 kHz. Scope connections to monitor the temperature gauge also are shown in Figure 3-4. It is advisable although not necessary to leave the oscilloscopes connected while demonstrating operation of the experimental model.



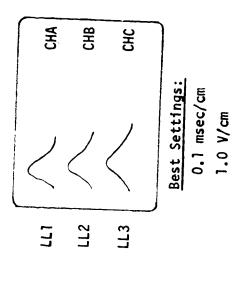


Figure 3-4. Oscilloscope Connections for Experimental-Model Electronics Calibration

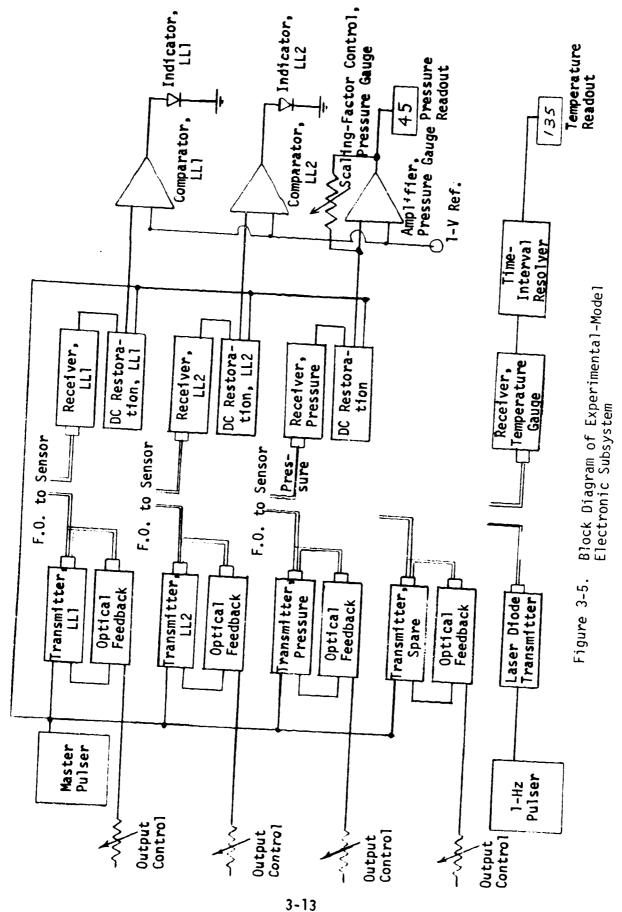
### 3.2.4 <u>Demonstration Procedure</u>

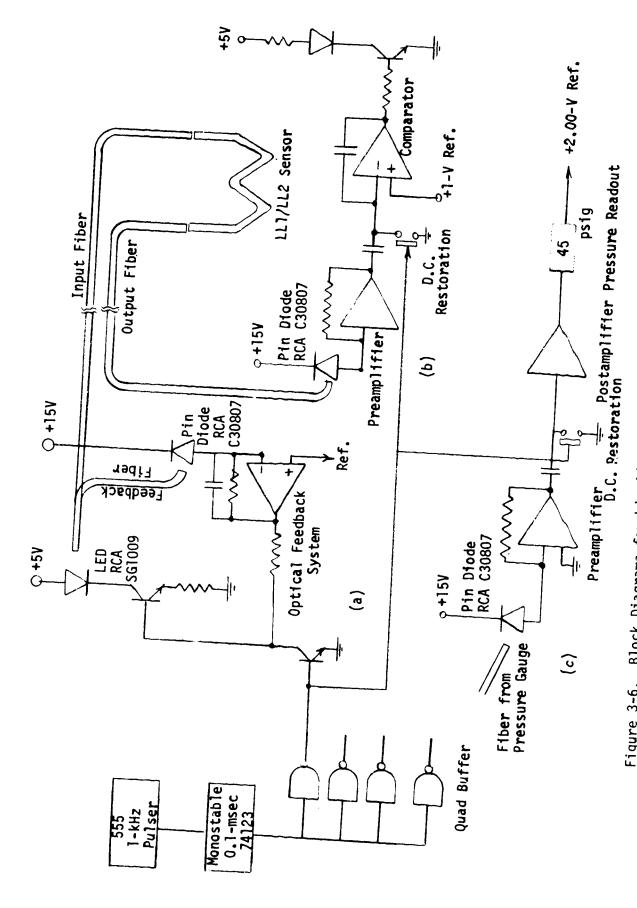
Liquid-level sensor operation is demonstrated by moving the sting assembly up and down (in and out) of the stabilized  ${\sf LN}_2$ . The sensing lights should come on when the sensors are immersed in the liquid.

The pressure sensor may be demonstrated by applying pressure at the  ${\sf LN}_2$  input line. This is most effectively accomplished by replacing the LN<sub>2</sub> bottle with a helium gas bottle. After closing the relief valve on the Dewar, pressure is applied by opening the valve of the helium gas container. Readings of the fiber-optic gauge are then compared with those of the mechanical pressure gauge at the helium bottle. Experimental data and a lookup table are included in Section 4. Temperature-gauge demonstration is accomplished by lowering the sting sufficiently to immerse the temperature sensor in the liquid helium. Because the temperature sensor is mounted approximately 1 inch higher than the No. 2liquid-level gauge, the sting therefore must be lowered 1 inch past the indication of that liquid-level sensor. At this point both the fiber optic sensor and the thermocouple should read 77°K. Operation of the temperature gauge can be demonstrated by raising the sensor assembly out of the liquid and observing the temperature rise. The two gauges should always agree with each other, although any thermal lag between the two can result in some difference in the readings.

### 3.3 ELECTRONIC SUBSYSTEM

Figure 3-5 is the block diagram of the total electronic subsystem of the experimental model. The electronics for the two level sensors and for the pressure sensor are located behind the main panel and the temperature-gauge electronics are located behind the other panel. Electronics for the two liquid-level sensors are shown in Figure 3-6(a). The master i-kHz pulser serves as an approximately 10-percent-duty-cycle pulser to all four transmitting LED drive circuits (i.e., both level sensors, pressure sensor, and spare channel). Each of the drive circuits apply power to its corresponding LED while also responding to the feedback circuit to keep the optical power of the LED at a constant level. The receive circuit of the level sensors is comprised of a





Block Diagrams for Liquid-Level, Pressure, and Temperature Gauges Figure 3-6.

high-gain low-noise transimpedence bandpass amplifier, a dc restorer circuit, a threshold detection circuit, and the indicating LED drive circuit. The block diagram is shown in Figure 3-6(b), and a complete schematic is included in Section 4. Although that block diagram details operation of the level-gauge electronics, a few specific characteristics particularly should be noted. For example, a pulsed signal rather than a dc signal was selected in order to minimize dc drifts in the receiver, and a 10-percent rather than a 50-percent duty cycle in order to obtain most peak power from the LED.

The pressure-gauge electronics as shown in Figure 3-6(c) obviously is very similar to the level-gauge electronics, except that the threshold-detection circuit has been replaced by a linear amplifier which drives the front-panel pressure readout.

Conversely, the temperature-gauge electronics differs greatly from the level-sensor electronics, as shown in Figure 3-7, in that the transmitter is comprised of a pulse generator that produces pulses at a repetition rate of approximately 1 Hz, with a pulsewidth of 10 microseconds. The pulse generator drives a 25-amp laser diode pulser which in turn drives a RCA SG 2007 laser diode. The optical power output of the associated laser is approximately 7 watts peak. The receive electronics also shown in Figure 3-7 is comprised of a low-noise high-gain wide bandwidth transimpedence preamplifier. The temperature-resolving electronics essentially is a time-interval-measuring circuit which measures the time between initiation of a laser power pulse and passage of the corresponding receiver pulse through the sensor. A detailed circuit diagram also is included in Section 4.

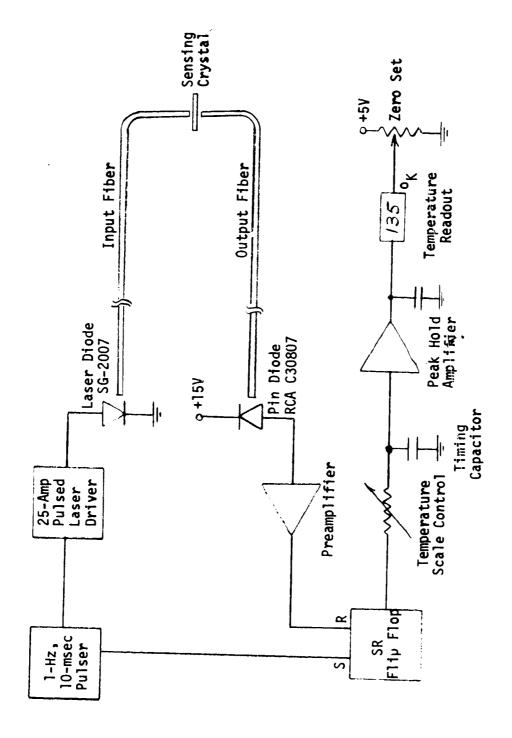


Figure 3-7. Block Diagram of Temperature-Sensor Resolving Electronics

# 4. SENSOR DEMONSTRATION MODEL DESIGN DESCRIPTION

This section summarizes design details, final data, and schematics of the TRW experimental cryogenic sensors demonstration model. Presented information reflects resolutions to problems recognized following other conceptual and sensor-selection phases of this project, during which it was found necessary to obtain various pertinent data that additionally were required for successful design of the final sensors.

Because the prevailing approach to level sensing proved in various ways to be inappropriate within cryogenic environments, it was found necessary to introduce many modifications in order to assure satisfactory level detection of cryogenic liquids having very refractive indices. Similarly, the pressure gauge required appropriate redesign to properly adapt it to its intended usage. On the other hand, the temperature gauge is essentially new, and several aspects of its operation had to be validated experimentally before its final design could be implemented.

### 4.1 LIOUID-LEVEL SENSOR

In the conceptual study the bare-fiber concept was selected. Although some of the preliminary experimentation clearly showed excellent feasibility (Ref. 2), these experiments were conducted with a HeNe laser without incorporating the signal-conveying fibers, fiber-optic connectors, and practical electronics. Consequently, practical level sensors based on the conceptual model suffered enormous loss, and the return optical signal was too small to be processed.

Several liquid-level sensors that were fabricated yielded the loss and sensitivity measurements itemized in Table 4-1. These data served to identify the approach to the final sensor shown in Figure 4-1. The new sensor has a throughput loss of 22 dB, sensitivity of 8 dB in LN<sub>2</sub>, calculated sensitivity of 2 dB in LH<sub>2</sub>, and a decision height of 0.25 mm. Further design work also was necessary to determine the most desirable adhesive for attaching the sensor to the sensor head. This was found to be "Scotchcast" XR-5241 material manufactured by 3-M Company, the only material that works satisfactorily in a cryogenic environment and that also has a high refractive index. This epoxy has been used to mount the sensor

Table 4-1. Liquid-Level Sensors Insertion-Loss and Sensitivity Data	s Insertion-Loss and S	ensitivity Data
LL Sensor Test Configuration	Insertion Loss (db)	Sensitivity (ab)
	0.3	N/A (LN <sub>2</sub> )
5	-36	6 (LN <sub>2</sub> )
	-32	4
	-10	æ.
	-18	ω

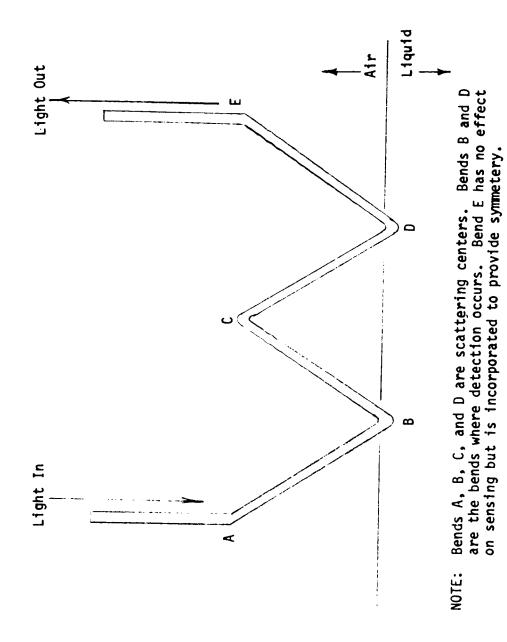


Figure 4-1. Final Liquid-Level Sensor

to the sensor chamber while providing a cryogenic seal between the liquid and the dry environment inside the sensor chamber. Because use of epoxy increases the throughput loss, such loss was avoided by applying a coating of platinum at the point where epoxy would be emplaced. Measurements of losses of various configurations are given in Table 4-2.

In the final design a special bushing has been used together with glass windows to minimize loss due to attachment of the sensor. The arrangement now being used, and also a newer technique which has not been incorporated in the present experimental model have been described in detail in a current TRW publication (Ref. 2). A schematic diagram of level-gauge electronics is presented in Figure 4-2.

### 4.2 PRESSURE GAUGE

The pressure gauge of the experimental model resembles the conceptual design, and no modification was found necessary. However, the present design manifests a thermal shock effect which results in a thermal drift, whose effect can be minimized using proper calibration techniques. As noted in the discussion of the conceptual phase, the pressure gauge is based upon the known technology of measuring the distance of a reflective diaphragm from a sensing bifurcated fiber bundle by determining the amount of light that is reflected through the bundle. Figure 4-3 evaluates the function of reflected light versus distance. It should be noted that this curve displays two slopes, which represent the behavior of the reflected light as the diaphragm moves away from the fiber. The reflected light increases up to a certain peak and then starts to decrease. The zero-pressure diaphragm placement is at the point indicated in Figure 4-3, and as pressure increases the reflected light decreases.

Pressure-gauge electronics are arranged in such a manner that a certain voltage level (1-V peak in this case) designates zero pressure. Any output voltage less than zero then is indicated on the meter as a pressure reading. In the experimental model the gain values of the pressure-gauge receive electronics translate to 1 millivolt per pound of pressure.

Table 4-2. Signal Loss Due to Epoxy

Sensor Configuration	Insertion Loss (dB)
(1 Epoxy Ring 1.5 mm Wide)	10
(3 Epoxy Rings Each 1.5 mm Wide)	14
(Platinum Coating 0.2 in Wide)	7
(Platinum Coating 0.2 in Wide Plus Epoxy Band)	6

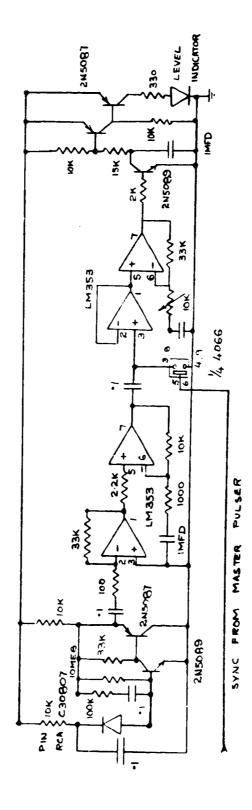
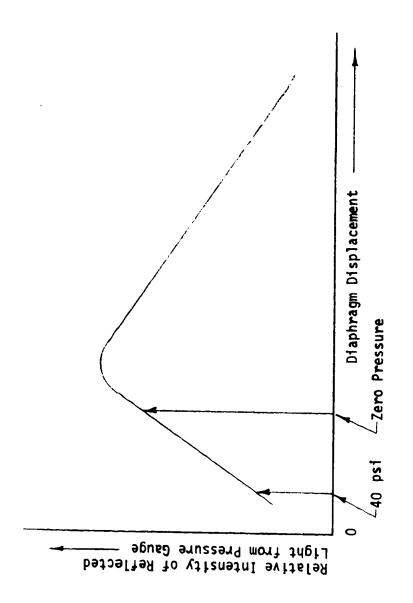


Figure 4-2. Level-Gauge Electronic Schematic



NOTE: Zero pressure adjustment shown at portion of curve yielding best linearity in pressure readings. A 0.006-mil diaphragm was selected to cover range of 0-40 psi.

Figure 4-3. Diaphragm Displacement vs. Reflected Light

Table 4-3 summarizes measured data up to 40 psi, within which range the pressure gauge is fairly linear. However, care must be taken while calibrating the pressure gauge in order to avoid effects of thermal shock. Table 4-4 is the associated lookup table. Pressure-gauge construction details and electronic schematics are shown in Figures 4-4 through 4-7.

#### 4.3 TEMPERATURE GAUGE

An InP crystal was selected as the sensing crystal to accommodate the desired temperature range of  $4^{\circ}K$  to  $135^{\circ}K$ . The bandgaps of InP and of GaAs are offset from each other roughly  $300^{\circ}C$  and the sensitivities of crystals operating at room temperature and at cryogenic temperatures are in the ratio of 1:10. Therefore, the temperature of the GaAs laser diode must be swept from room temperature to  $25^{\circ}$  above room temperature to cover the desired range of temperature measurement.

The basic premise of temperature measurement for present purposes is to measure bandedge shift. Also, the key issue in designing the temperature gauge is to develop simpler methods of measurement. Some of the suggested methods incorporated use of complicated and delicate optical components, and the effect of bandedge shift was measured in terms of amplitude. However, measurement of amplitude always includes drifts that cause inaccuracies in measurement. Also, resolution of the prevailing method is poor because the slope of the bandedge curve from 10 to 90 percent of the slope is  $55^{\circ}$ C wide, as shown in Figure 4-8 for a GaAs crystal. The swept-spectrum method of measuring bandedge shift has been developed and incorporated in the experimental model. Full description of this technique has previously been published (Ref. 5). The light source is an RCA SG 2007 pulse laser diode. Although several other diodes also were considered, this wavelength and the thermal time constants of the SG 2007 diode were most suitable for this experiment. An elaborate setup including a spectrophotometer was used to measure the emitted wavelength shift as well as its range. Figure 4-9 presents oscillograms of this experiment, with results showing that the output wavelength of an SG 2007 diode shifts by 130 Å within 10 microseconds of the applied pulse.

Table 4-3. Pressure-Gauge Calibration Readings for Gauge Operating in  ${\rm LN}_2$ 

Pressure-Gauge Lookup Table for Gauge Operating in LN<sub>2</sub>

Table 4-4.

1,444.

5 - 10 mm

T. Sales .

Reading	00	20	1	20	29	38	56
Pressure (psig)	0	ડ	10	20	30	40	09

Pressure (psig) 10 20 8 g 33 41 **6**8 Reading 0 20 2 10 30 40 9

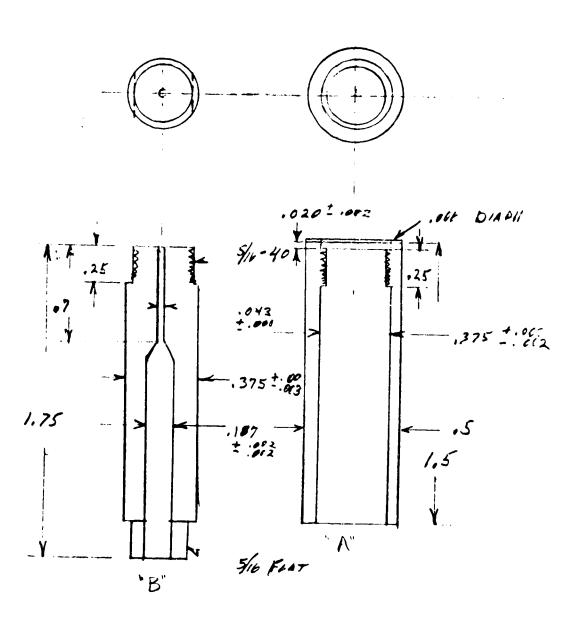


Figure 4-4. Pressure-Gauge Construction Details

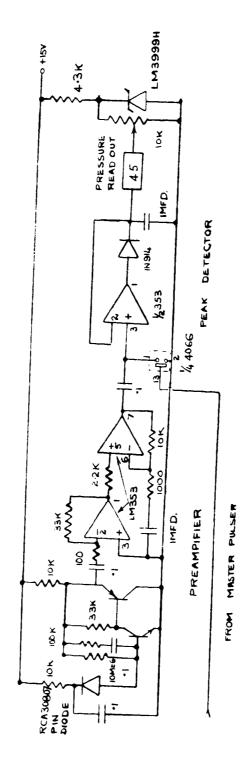


Figure 4-5. Pressure-Gauge Electronic Schematic

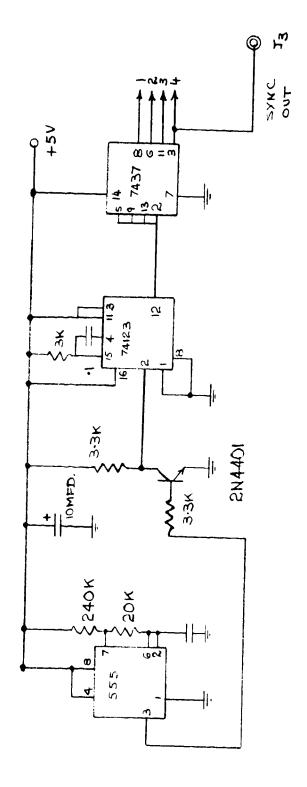


Figure 4-6. Schematic of Master Oscillator

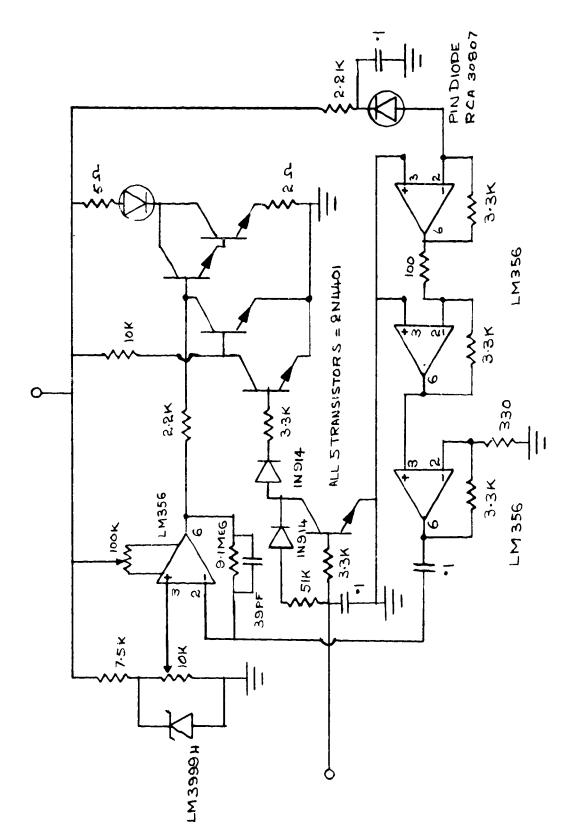


Figure 4-7. Schematic of LED Drivers with Feedback Control

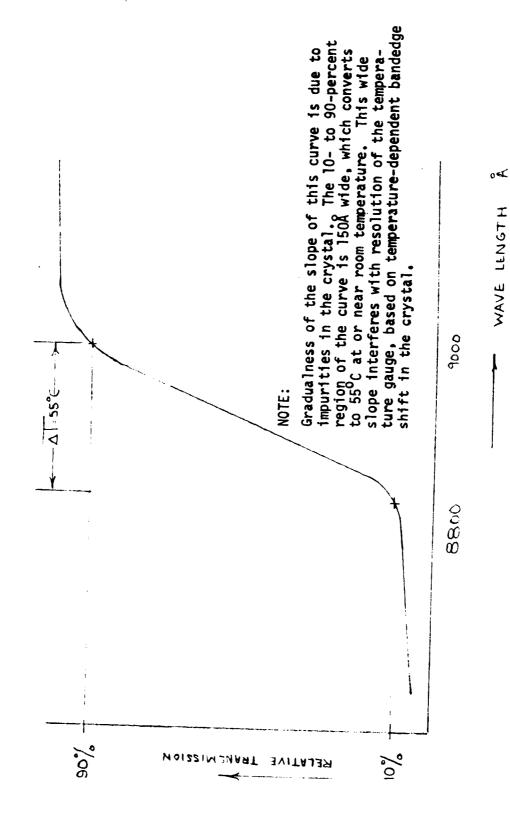
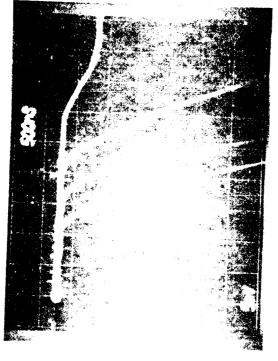
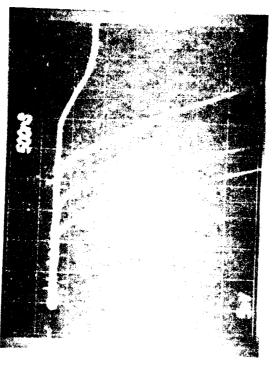


Figure 4-8. Bandedge Curve of GaAs Crystal



(a) Emission with Laser Diode at Room Temperature



(b) Emission with Laser Diode at  $-30^{\rm O}{\rm C}$ 

Figure 4-9. Oscillograms Showing Bandedge vs. Temperature for GaAs Laser Diode

Figures 4-9(a) and 4-9(b) illustrate emission wavelengths of an RCA SC-2007 laser diode at various junction temperatures. These exposures were obtained while the laser was being energized with 3-µsec 25-Amp pulses. The light was monitored through a monochromator tuned at various wavelengths. The time delay between initiation of a laser pulse and appearance of light is the time required for the laser-diode junction to heat sufficiently to emit wavelengths to which the monochromator is tuned. In Figure 4-9(a) the laser diode at room temperature has generated five wavelengths of 9050Å, 9060Å, 9070Å, 9080Å, and 9090Å. Figure 4-9(b) shows results of the same experiment with the laser junction cooled to -30°C, producing wavelengths from left to right of 8970Å, 8980Å, 8990Å, and 9000Å.

These data were used to design required temperature-gauge electronics, for which a block diagram is shown in Figure 4-10. The transmitted pulse is 10 microseconds long and has a repetition rate of 1 Hz. Temperature is measured in terms of the time interval between initiation of the laser pulse and pulse arrival at the sensing crystal. The temperature gauge indicates the  $77^{\circ}$ K (LN<sub>2</sub> temperature) reading by a time interval of approximately 4.5 microseconds.

Obtained experimental data of temperature versus temperature-gauge readings and the associated lookup table are given in Tables 4-5 and 4-6. Mechanical arrangement of the temperature gauge is depicted in Figure 4-11. Electronic schematics of the temperature gauge and of the associated 25-Amp laser-diode pulser are given in Figures 4-12 and 4-13 respectively.

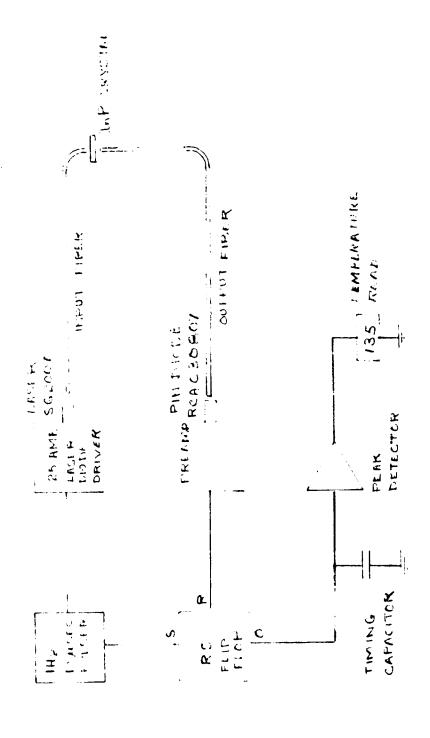


Figure 4-10. Block Diagram of Temperature-Gauge Electronics

Temperature-Gauge Calibration Readings Table 4-6. Temperature-Gauge Lookup Table for Gauge for Gauge for Gauge for Gauge Manage operating in LN  $_{
m 2}$ Table 4-5.

( <sup>9</sup> K) 78 81 89 96 99 112 115 130 170
81 89 96 99 112 115 130 170
89 96 99 112 130 150
89 96 99 112 115 150
96 99 112 130 150
99 112 115 130 150
112 115 130 150
115 130 150 170
130 150 170
150
170

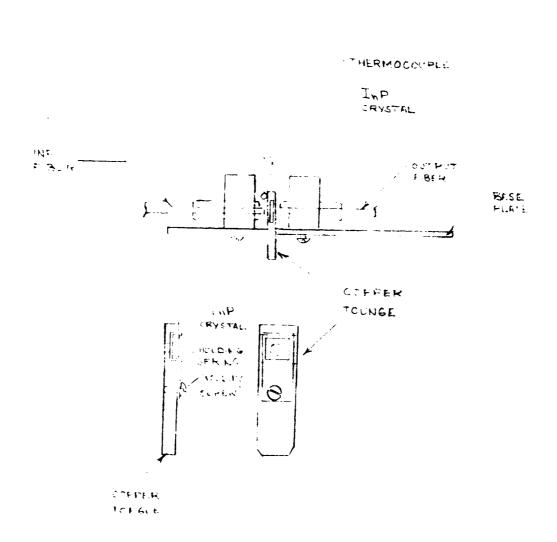


Figure 4-11. Temperature-Gauge Mechanical Arrangement

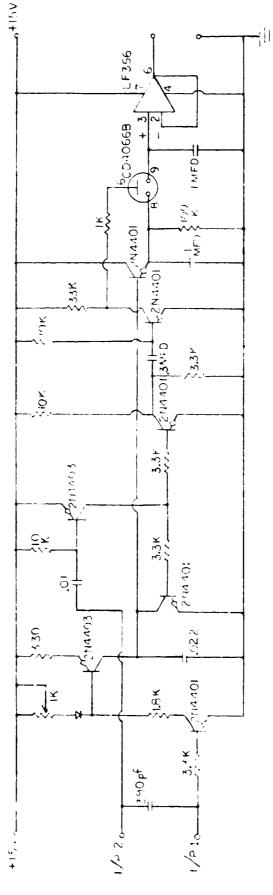


Figure 4-12. Temperature-Gauge Electronic Schematic

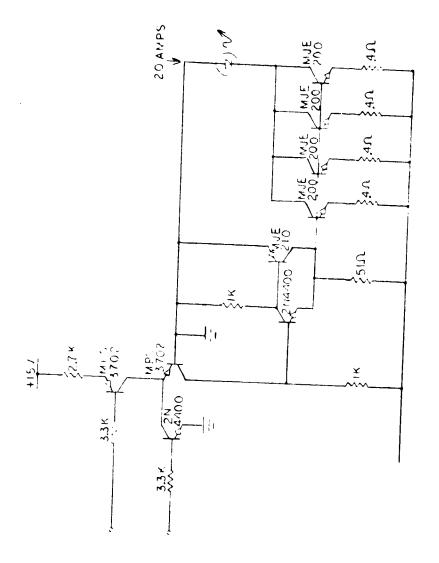


Figure 4-13. 25-Amp Laser-Diode Pulser Electronic Schematic

#### REFERENCES

To facilitate most convenient review, facsimile copies of the following references also are included among the technical material presented in Appendix A of this report.

- 1. "Principles of Fiber-Optic Systems." Informal unnumbered description of fiber-optic system elements and technology. TRW/DSSG, One Space Park, Redondo Beach, California.
- 2. "Fiber-Optic Liquid-Level Sensor New Technology Report." TRW patent description Report No. 11-182, TRW/DSSG, One Space Park, Redondo Beach, California (17 Oct 1979).
- 3. "Test Plan and Procedures for Fiber-Optic Instrumentation." Report No. 31262-6002-RU-00 prepared for NASA/JSC, TRW/DSSG, One Space Park, Redondo Beach, California.
- 4. "The Bandgap and the Temperature Coefficient of the Bandgap of GaAs Between 0° and 300° K." TRW/DSSG Memorandum 78.4351.2-021, TRW/DSSG, One Space Park, Redondo Beach, California (10 Feb 1978).
- 5. "Swept Spectrum Temperature Gauge." TRW/DSSG Memorandum 78.4351.8-013, TRW/DSSG, One Space Park, Redondo Beach, California (9 Feb 1979).

## APPENDIX A

## CONTENTS

1.	Reference 1	Facsimile											ē	Α-
2.	Reference 2	Facsimile			•			•		•				A - 2
3.	Reference 3	Facsimile											•	
4.	Reference 4	Facsimile	•						_		•	•	•	A-4
5.	Reference 5	Facsimile							•	•	•	•	•	
6.	Useful Desig	n Data .	•	•				·	•	•	•	•	•	A-5
						-	-	•	•			_		u _ F

# PRINCIPLES OF FIBER OPTIC SYSTEMS

# Basic Instrumentation Elements

The basic elements of any fiber optic instrument system are as shown in Figure 1. The system functions more or less as a communication link. Optical signals from a light source are transmitted via a fiber optic cable to a remote sensor which then "conditions" the transmitted light beam according to the measurement being made (in our case, the temperature). The resulting, modified light output from the sensor is returned to the optical receiver through another fiber optic cable, the signal output of the receiver thus furnishing the information about the sensed conditions

In the following sections, we present some detailed discussion of the principles of optical fibers and the associated components of a fiber optic system. Some of the material is more general in nature than required for the instrument and control systems of concern here, but is included as an overview of the basic principles.

# The Source: LED's or Avalanche Diodes

The light source (or emitter) for the system must be capable of modulation at a rate sufficiently fast to handle the information flow. It can be externally modulated by a fast light shutter or light modulator, but this is usually much more complicated and costly than controlling the intensity directly by means fo the current flow. Second, the wavelength of the light from the source must be within a part of the spectrum that allows efficient transmission through the fiber. Third, the emitter area should be such that sufficient optical power can be coupled into the small diameter optical fiber or bundle. This last requirement rules out most conventional light sources, such as incandescnet lamps, where the light is dispersed over a large area. Thus the source must have high radiance and a small effective source diamter; otherwise, most of the light will be lost.

There are two kinds of light sources compatible with optical fibers: light emitting diodes (LED's) and the closely related injection laser. Both LED's and injection lasers are solid state devices, and can be made small enough to work well with fiber optic cables; in addition, their unit costs are typically low, of the order of a few dollars.

#### Light Emitting Diodes

LED's for fiber optic communications are p-n diodes made from crystals of aluminum-gallium-arsenide (AlGaAs). Forward biasing of the junction causes the injection of electrons from the n-region where they recombine with holes, causing photons of light to be emitted. The emitted light is incoherent (random phase) and isotropic (nondirectional) and has a lower radiance than that from injection lasers. LED's not specifically designed for fiber optic use, e.g., display and sensor applications, generally have a low radiance and at best work only with fiber bundles that offer a large cross-section (typically 0.045 inch diameter) to capture the light.

For single fiber applications, higher radiance is needed and special devices have been designed for the purpose. Figure 2 shows one of these, the Burrus diode, named for its invetor at Bell Laboratories. This LED has low thermal resistance, high current density, and provides a "well" for the optical fiber so that coupling losses are minimized. Its useful power output is typically about one-tenth that of an injection laser but is adequate for many fiber-optic applications.

LED's can be directly modulated at rates up to a few hundred megahertz, and they have fairly linear power-drive characteristics, making them useful for analog applications. At the present time they also are less expensive, more reliable, and have a longer operating lifetime than injection lasers. Their big disadvantage is their lower radiance, which means lower coupled power and more repeater stations in long-haul fiber optic communication links.

#### Injection Lasers

Injection lasers are similar to LED's in that they use the same material, AlGaAs, to form a p-n junction. The light producing mechanism is also the same; however, in the injection laser the p-n junction is perpendicular to the two cleaved ends of the crystal which form the reflecting mirrors of an optical cavity. Laser action occurs when the optical gain in the recombination region exceeds cavity losses. The emitted light is therefore coherent (in phase) and directed, and has a high radiance. The spectral distribution is narrower than that of the LED, about 0.002  $\mu m$ .

Laser action in this type of device was first achieved in 1962, about two years after the first ruby laser. These early minature semiconductor lasers were simple p-n junctions (so-called "homojunctions" since a single material--gallium-arsenide--was used. To trigger the laser action, extremely high current densities were required (typically 50,000 amperes per square centimeter), so that they could only be operated in a pulse mode at reduced temperatures (liquid nitrogen). Even then, their operating lifetimes were measured in minutes.

Fortunately for fiber optics applications, significant advances have been made, and continuous injection laser operation at room temperature is now common. The development that made this possible is the double heterostructure (DH) injection laser shown in Figure 3. In this device, the thin recombination region is bounded by two heterojunctions, p-p and n-n, formed by changing the aluminumgallium ratio. These heterojunctions serve two functions. First, waveguiding of the radiation occurs because the recombination region is bounced by material with a lower refractive index. Second, replacing gallium by aluminum increases the energy band gap outside the recombination region, causing minority carrier containment within the recombination region. Also, because the active region is confined to a narrow strip, typically about 13  $\mu\text{m}$  wide, the device provides a high concentration of injected carriers (high optical gain). The required current density is greatly reduced--to a few hundered amperes per square centimeter-and the threshold current to cause laser action is typically 100 mA. By proper choice of the aluminum-gallium ratio, the recombination band gap and therefore the wavelength of the emitted radiation can be varied.

### The Medium: Optical Fibers and Cables

The basic fiber lightguide consists of a transparent core surrounded by a cladding material having a lower refractive index. The refractive index of quartz is 1.46 and for common types of glass ranges up to about 1.7. When light travels from a high index region to a low index retion at a sufficiently shallow angle (termed the critical angle for total internal reflection), it is completely reflected; at a steeper angle the light rays are partially transmitted into the low index region.

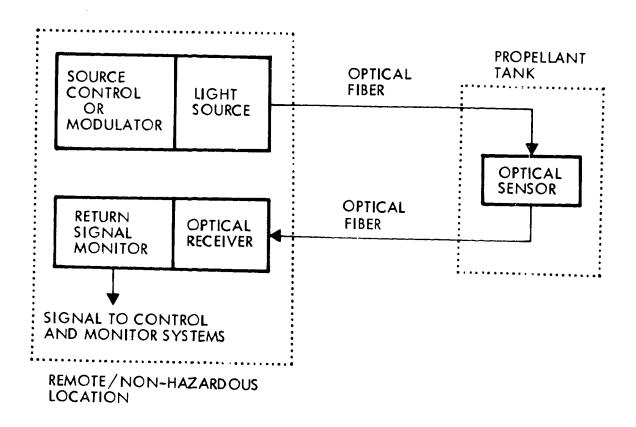


Figure 1. Basic elements of a fiber optic instrument system for remote measurements of propellant tank conditions.

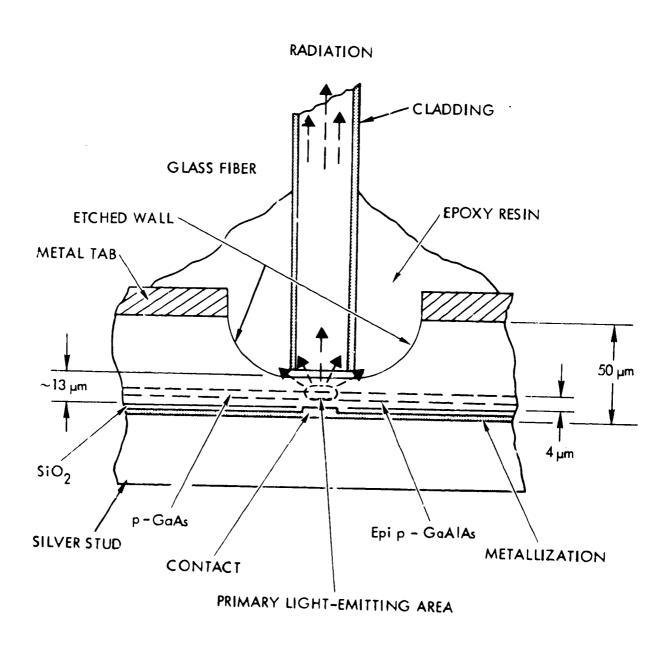


Figure 2. Construction of a Light-Emitting Diode (LED).

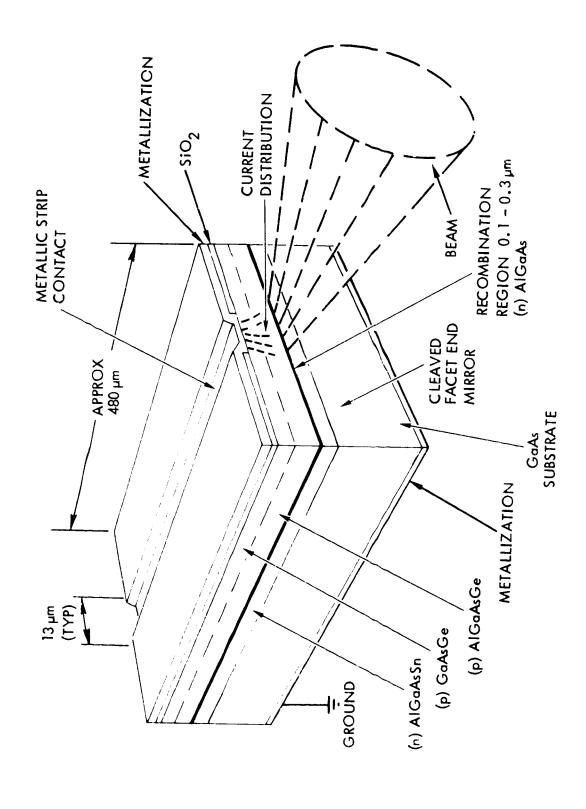


Figure 3. Double Heterostructure Injection Laser.

Figure 4. Cross-section of fiber optic waveguide with reflected light rays.

The simplest concept of light-guiding in an optical fiber is to visualize the light at being totally internally reflected by the core-cladding interface. In more detailed terms, the fiber acts as a dielectric waveguide which supports only a limited number of propagation modes. The modes (or eigenfunctions) are solutions to the wave equation, with boundary conditions dictated by the structure through which the energy propagates--in our case the optical fiber. There are guided modes and unguided or radiation modes. We are most interested in the guided modes since they are confined by the fiber and travel with minimum loss down it.

For a fiber waveguide of finite size, there are a finite number of guided wave modes for a given optical frequency that is a finite number of discrete solutions to the wave equation. Each mode exhibits a unique spatial distribution of energy across the fiber. Light guided by the fiber can be decomposed into the various permitted modes. The amount of energy in each mode depends on the optical emitter, coupling mechanisms, and irregularities in the fiber that can cause mode conversion. With smaller fibers, fewer modes are allowed. The single mode fiber, as its name implies, propagates only one mode.

An important characteristic of an optical fiber is its numerical aperture (N.A.), defined as the sine of the maximum half-angle of acceptance ( $\theta$ ) of the incoming light beam. Light entering the fiber at a larger angle will leak through the core-cladding interface and be lost. Thus, the theoretical extremes for the value of N.A. are 0 and 1, corresponding to half-angles of acdeptance of 0 and 90 degrees, respectively. When maximum light acceptance (large number of propagation modes) is important, high N.A. fibers are used, and values of 0.7 or higher are not uncommon. For fiber optic communications, lower values (typically 0.1 to 0.2) are used, since a large number of propagation modes leads to problems in modal dispersion, which produces pulse smearing in digital systems and frequency roll-off in analog systems, because different modes travel at different relative propagation velocities through the fiber. Figure 4 shows the relationship between the half-angle of acceptance and the refractive indices of core and cladding.

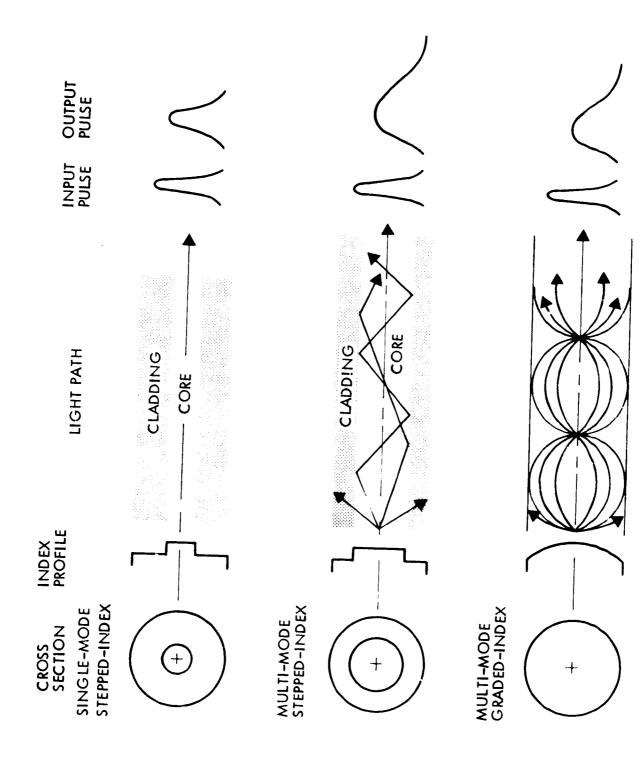


Figure 5. Three types of optical fibers.

# FIBER OPTIC LIQUID LEVEL SENSOR NEW TECHNOLOGY REPORT

#### PROBLEM

At least 12 patents have been applied for the bare fiber liquid level, most of these innovations pertain to: sensing of ordinary liquids such as water, gasoline and oil-liquids with refractive indeces much higher than that of the air and, operable at or near room temperature. To design the sensors for the low refractive indeces that operate at cryogenic temperature several new techniques have been employed.

The bare fiber sensors work on the principle that the optical throughput loss is dependent upon the material that surrounds the bare fiber. A practical sensor, therefore, is a glass rod connected to the fiber optic system that continuously monitors the throughput loss. When the sensor is inumersed in the liquid, the loss increases indicating that the sensor is in contact with the liquid. A typical system is shown in Figure 1. Figure 2 shows some of the existing sensors; multiple bends are to increase the sensitivity.

For most efficient sensors, two factors are important: the change in the throughput loss due to the refractive index of the liquid (sensitivity) and, the insertion loss of the sensor. Usually these are opposing factors since the sensitivity can be increased by adding bends to the sensor which also increase the insertion loss.

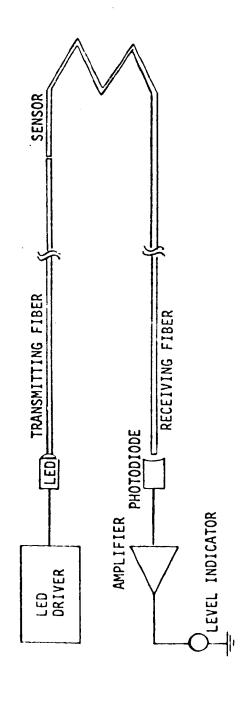
TRW SYSTEMS CERTAIN

PATENT DEFAMILLAT

Madan Sharma Date C. Roychoudhuri Date

R. E. Brooks Date

A2-1



Shows the schematic of liquid level sensing system. The LED emits 200 micro second wide pulses at a frequency of 50 Hz. The receiver derives its timing signal from the transmitter for synchronous detection. Figure 1.

Madan Sharma

Date

C. Roychoudhuri
A2-2

Collegia Date

R. E. Brooks

11-182

Shows typical sensors. Premis used in designing these sensors is that the lineal attenuation changes when the sensor is immersed in the liquid. Figure 2.

TRAY STOTE OF A STORY OF THE ST

10/16/19

Madan Sharma Date C. Roychoudhuri Date

. **F**.

R. E. Brooks

During initial evaluation of the sensor with techniques available at that time, it was found that to have a sensitivity of 2 dB for LH<sub>2</sub> would require 5 bends. The insertion loss of such a sensor was found to be 45 dB. This loss was enormous. Even using fairly expensive LED's the ouput of the sensor was too low to be measured. CW lasers and pulsed lasers were considered. CW lasers have not been tried. The pulsed lasers exhibited amplitude jitter and interferred with decision making threshold electronics.

#### **CBJECTIVES**

There are two problems to be considered: increase the sensitivity while keeping insertion loss down to 20-25 dB. and explore different means of attaching sensors to the fibers while maintaining a good pressure and liquid seal between the sensors and the signal conveying fibers.

#### **PROCEDURE**

To design more appropriate sensors, the phenomenon that does the liquid sensing must be examined. As often misquoted, the lineal attenuation of the glass does not change when immersed in the liquid, it is the numerical aperture which changes at the point of contact with the liquid. This can be explained with the aid of Figure 3. When the fiber is immersed in the liquid it is equvalent to two fibers. The one in the air has large exit numerical apertue while the portion in the liquid has small NA, since NA depends upon the refractive index of the surrounding material. The extra loss due to mismatch of the numerical aperture is in fact the sensing mechanism. This discussion clearly shows that the sensing is only at the surface of the liquid; no TRIV SYSTEMS & FATENT DEPARTMENT DEPARTMENT

Madan Sharma Date C. Roychoudhuri Date R. E. Brooks

8

The liquid sensing is due to N.A. mismatch between the sections of fiber in the air and the section in the liquid. Figure 3.

That & Street

Madan Sharma Dai Date

C. Roychoudhuri
A2-5

Date

R. E. Brooks

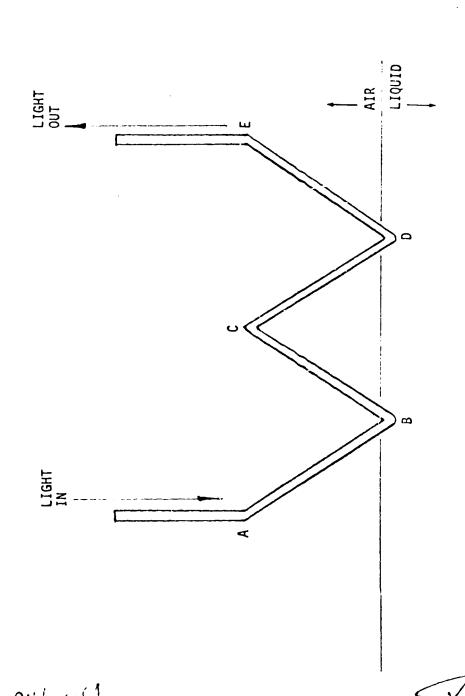
Another point worth considering is the contents of variety of optical modes present in the sensor. The loss due to mismatch of numerical aperture is dependent on the presence of oblique modes. An optical flux flowing through a long fiber sheds most of its modes along the length of the fiber. If well behaved modes are predominant the loss due to NA mismatch would be too small to detect - less sensitivity.

This discussion also points out that mismatch loss occurs only when the sensor goes from air to liquid but no sensing is done on its return trip if the sensor is u shaped. Mounting of the sensor is yet another problem. The configuration of this experimental system nessitates liquid and pressure cryogenic seal between the sensors and the signal conveying fiber optic cable. There are very few epoxies that are fit for this purpose. These epoxies have yery high refractive index, any application of this results into heavy throughput loss.

Such constraints are the reason for applying new technology to the level sensors to be used in cryogenic environments when refractive index of the liquid not much higher than air.

#### **PROGRESS**

New liquid level sensors have been designed. Figure 4 shows the sensor being used in the experimental set-up. The total insertion loss is 20 dB and sensitivity is 8 dB in liquid nitrogen which means 2 dB in LH2. The basic design approach was to achieve sensitivity with less number of bends. This was achieved by introducing scattering centers with sharp bends. A curve fiber has no good effect on the sensor, therefore, the legs of the sensor are kept straight. As has been discussed that the sensing loss is provided AIENI DEFALMENT by a single point where the sensor touches the liquid, the final sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the presence of liquid at its lowest points. The deep Part of the legs of the sensor detects the liquid at legs of the sensor detects the legs of the s



Shows the new sensor. The 5 bends A, B, C, D, are scattering centers B & D are bends where detection takes place. The bend E has no effect on sensing. It is there to provide symmetry.

Figure 5. FAIFAT B. c.

Madan Sharma

Date

11/16

Charcher C. C. Roychoudhuri
A2-7

cdy 71 Date

R. E. Brooks

10/15/19
Date

# 11-182

therefore, is less than the diameter of the bare fiber which is in this case one millimeter.

The sensor is sealed with the appropriate epoxy using a special bushing which ensures a very fine band of point of contact with the sensor. The bushing also ensures that the seal is very near to the end of the sensor.

Figure 5 shows a modified sensor sealing arrangement. This method is being studied presently.

Altoaulice 10/16
Madan Sharma Date

e G. Roychou

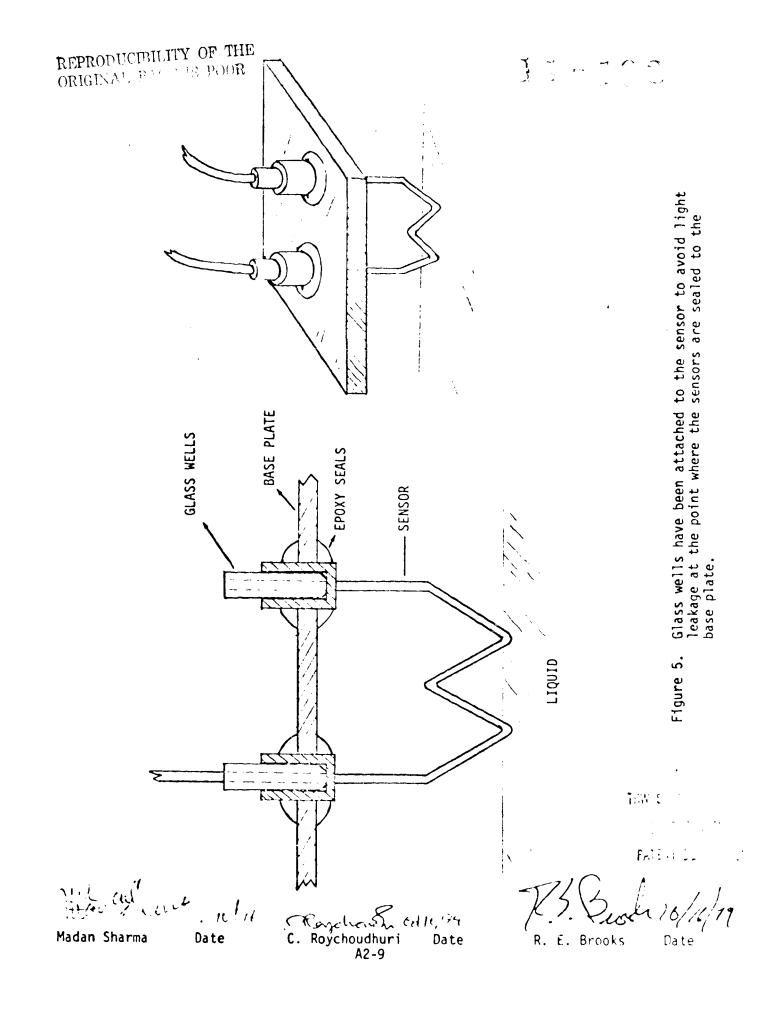
Ccl.1; 79

4.7. Vi

10/16/19

Date

TRIV SYSTEMS & EMERGY
OCT 17 1979
PATENT DEPARTMENT



#### "TEST PLAN AND PROCEDURES FOR FIBER OPTIC INSTRUMENTATION"

by

Ronald Watson

Prepared for Contract NAS9-15454 NASA Lyndon B. Johnson Space Center

Approved by: R8. Wille

R.S. Witte, Manager TRW Electro-Optics

Department

Tack D. Alexander Technical Manager

NASA/JSC

TRW Defense and Space Systems Group
One Space Park Redondo Beach, California 90278

#### 1.0 INTRODUCTION

The purpose of this document is to present the plan and procedures for testing the optical sensors developed for NASA Contract NAS9-15454. These sensors utilize fiber optic technology to provide passive sensing of liquid level, temperature, and pressure in cryogenic liquid propellant tanks.

The end product of the program is to be an engineering model, deliverable to NASA/JSC, with a set of working sensors to measure the level, temperature, and pressure in liquid nitrogen in a stainless steel (pressurizable) Dewar. Electronics components will be placed outside the LN2 tank, with measurements indicated on an instrument panel. Details of the program and sensor concepts are given in Reference 1.

Because the model tests do not involve operation in hazardous environments and are conducted at pressures below 15 psig (see Sction 2.1 below), the test plan and procedures outlined here involve no special requirements beyond standard laboratory research methods.

#### 2.0 TEST PLAN

#### 2.1 Test Requirements

The sensor performance goals are given in Table 1. Note that many of the goals listed are design goals only, associated with hazardous operation in LH2 and LOX and at high pressures; these design goals are not subject to test or demonstration on this program.

The tests to satisfy achievement of the demonstratable performance goals require insertion of the 3-sensor set into a quiescent bath of LN2 in a Dewar to contain no higher than 15 psig.

<sup>\*</sup> Taken from the contractual statement of work, as amended February 23, 1979.

#### Table 1. Sensor Performance Goals

ENVIRONMENT: Demonstrated operation in 77°K LN2 (system

to be designed for LH2 and LOX operation").

OPERATION MODE: Demonstrated operation for LN2 covering

and uncovering the sensors (dry-to-wet

and wet-to-dry operation).

LEVEL ACCURACY: Designed to obtain 0.1% by volume of the

Shuttle tanks.

TEMPERATURE RANGE: Demonstrated measurements in 77°K (LN2)

range (designed for measurements to 200K/LH2

and survival to 150°F/launch-stand con-

ditions\*).

PRESSURE RANGE: Demonstrated measurements in 0-15 psig range

in LN2 (designed for measurements to

300 psig\*).

TIME RESPONSE: Demonstrated transient response in LN2 as

follows: 50-100 msec for level sensor, 500 msec for temperature sensor (pressure sensor designed for 50-100 msec response).

SENSOR INSTALLATION

ACCURACY: Designed for ± 0.1 inch.

SENSOR LOCATION: Demonstrated capability up to 100 feet

from electronics.

FLOW LOCATION: Designed to withstand up to 27 fps.

LIFETIME: Demonstrated to withstand over 200 cycles.

<sup>\*</sup> These design goals can be satisfied by appropriate analyses and are not subject to test or demonstration on this program.

### 2.2 Test Hardware Configuration

The test hardware configuration shall consist of the following elements:

- Sensors (to measure level, temperature, and pressure)
   mounted on an extendable sting to insert into the LN2
   bath,
- A stainless steel, pressurizable Dewar to contain the LN2, and the cover of which will support the sensor mounting sting,
- Associated electronics containing power supplies, light sources, and measuring and indicating systems,
- A console to contain all of the above.

## 2.3 Facilities and Test Equipment Requirements

The test facility is the LN2 Dewar, self-contained on the engineering model console, capable of maintaining the LN2 for a period of approximately four hours. Standard 110 V AC electrical power is required to operate the instrument electronics.

### 2.4 Test Schedule

The tests of the engineering model shall take place in the period of 15 March 1979 to approximately 6 April 1979.

## 2.5 Consumable Materials

The consumable materials are approximately 120 gr of LN2  $(15 \ liters)$ .

# 2.6 Minimum Quality Assurance and Reliability Requirements

Electronic and pressure indicating devices shall be calibrated as per TRW equipment pool procedures.

#### 2.7 Special Safety Precautions

Standard procedures for handling LN2 shall be applied.

Special automatic and manual pressure relief vent valves shall be provided to limit pressure in the Dewar to 15 psig.

#### 2.8 Minimum Test Procedure Requirements

The minimum test procedures are that the LN2 level will have been stabilized at atmospheric pressure before insertion of the 3-sensor set. The exhaust vent shall be set to relieve pressure from LN2 boil-off at the 15 psig level.

# 2.9 Special Technical and Engineering Support Requirements None.

#### 3.0 TEST PROCEDURES

#### 3.1 Scope and Description of the Tests

The tests shall be the measurements of the response of the 3-sensor set during immersion into and withdrawal from the LN2 bath. The static and transient response behavior of the instruments will be measured up to 15 psig.

#### 3.2 Safety Precautions

The test personnel will insure that the 15 psig pressure relief vent is in working order as the test proceeds. Furthermore, the pressure gauge shall be monitored at all times to insure that the pressure does not exceed 15 psig in the Dewar; a manual relief vent is available to override the automatic vent if necessary to drop the pressure.

### 3.3 Test Preparation and Checkout Instructions

The Dewar shall be filled with LN2 to one-half to two-thirds full, and maintained for a time period to insure a quiesent

liquid surface as determined by visual inspection with the instrument mounting system and Dewar cap removed. Before installation of the instrument mounting system in the Dewar, the electronic subsystems shall be checked to insure proper light source operation and optical continuity through the fiber bundles and connectors.

#### 3.4 Test Procedure

After filling and equilibration of the LN2 Dewar as per the Test Preparation instructions in item 3.3 above, the instrument assembly shall be mounted in the Dewar with the sensors in the "up" position, above and not immersed in the LN2. Monitoring of the sensor output signals shall proceed at this time to determine behavior during the cool down period.

After temperature equilibration of the sensors and mounting box assembly in the non-immersed position, as determined by stable output signals, the sensor outputs shall be monitored during a controlled rate of insertion into the LN2, so as to determine immersion time response characteristics.

After insertion of the sensors in the LN2 and further temperature equilibration in the immersed position, static response characteristics shall be measured to determine the required calibrations. Quasi-static response of the pressure sensor shall be obtained by relating the sensor output to the pressure in the Dewar as determined from the reference pressure gauge which monitors the Dewar tank pressure. Temperature calibration shall be obtained from the known temperature-pressure characteristics of LN2 (77° K at 1 atm) plus a reference thermocouple attached to the sensor holder. Static response of the level sensor is determined by the on-off characteristics when above or immersed in the LN2.

After static, immersed response measurements have been

obtained, the sensors shall be removed from the LN2 at a controlled rate to determine the transient response characteristics for this mode of operation.

The capability of the sensors to undergo a lifetime of 200 cycles shall be determined by cycling the sensors in and out of the LN2 200 times with the immersion device.

3.5 Emergency Shutdown or Special Hazardous Security Procedures

The manual vent relief value shall be opened if the Dewar tank pressure exceeds 15 psig.

- 3.6 Requirements of Quality Assurance of Compliance with Mandatory Procedural Instructions

  None.
- 4.0 DESIGN OF SPECIAL TEST EQUIPMENT, FACILITY EQUIPMENT, AND TEST INSTALLATIONS

The Dewar test tank is the only item of equipment constructed especially for the tests described herein. This tank was designed and fabricated by Andonian Cryogenics, Inc. of Newtonville, MA, to operate down to liquid helium temperatures (4.2°K) and up to 300 psi (20 atm) pressure; the tank is a high pressure version of Andonian Model number SCSM-3-1/2 with a capacity of 3.3 liters at 50% full. Figure 1 is a sketch illustrating the Dewar construction.

- 5.0 LIST OF REFERENCES
- 1. R. Watson "Sensor Recommendations for Fiber Optic Instrumentation", 31262-6001-RU-00, TRW DSSG, 31 March 1978.

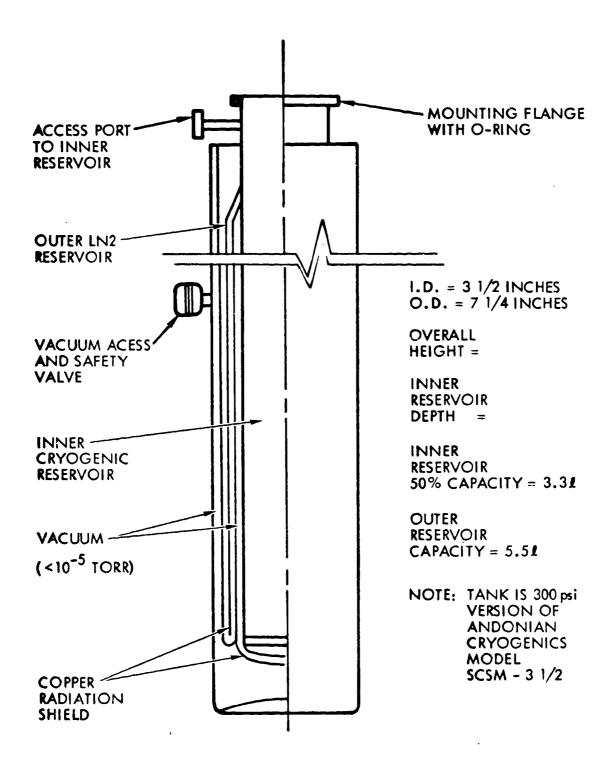


Figure 1. Special Stainless Steel Dewar Tank

INTEROFFICE CORRESPONDENCE

78.4351.2-021

TO:

R. Brooks

M. Sharma

R. Watson R. Witte

CC:

DATE

10 February 1978

The Bandgap and the Temperature Coefficient of the Bandgap of GaAs

Between 0 and 300°K.

FROM:

W. Von der Ohe

BLDG MAIL STA.

EXT.

R1

1062

61816

REFERENCE: J. Pankove, Optical Processes in Semiconductors, Prentice Hall, Englewood Cliffs, New Jersey, 1971, p. 27:

The tempera-

ture dependence of the gap for many semiconductors has been fitted by the following empirical relation:17

$$E_{\mathfrak{s}}(T) = E_{\mathfrak{s}}(0) - \frac{\alpha T^2}{T + \beta} \tag{2-2}$$

where  $E_a(0)$  is the value of the energy gap at  $0^{\circ}$ K and  $\alpha$  and  $\beta$  are constants. Table 2-3 lists these values for several semiconductors.

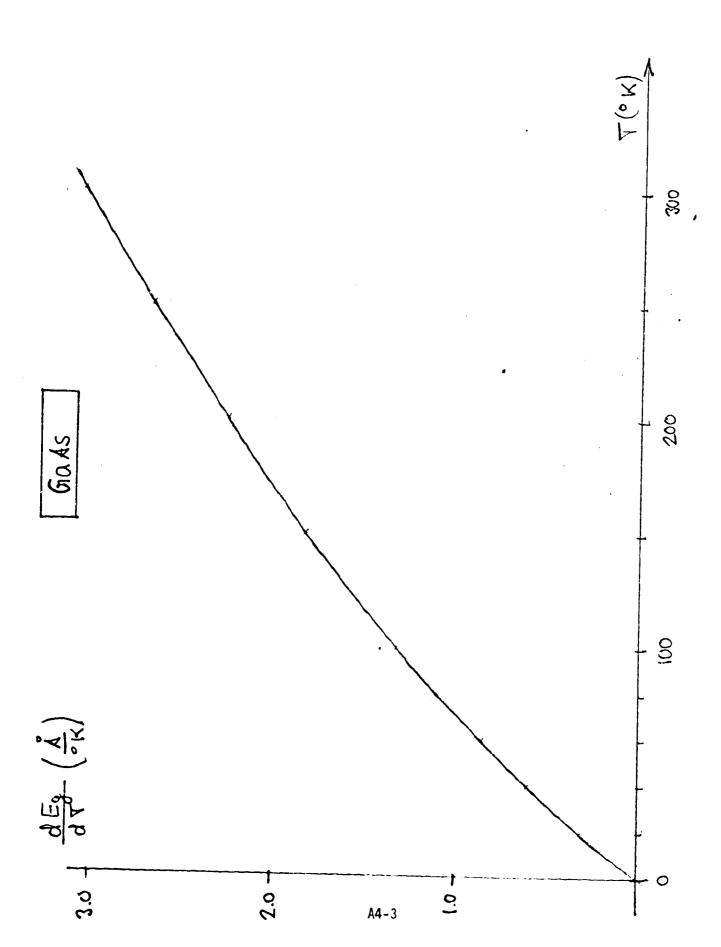
Table 2-317 Values of the parameters in Eq. (2-2)

Substance	Type of gap	$E_{\mathbf{g}}(0)$ (cV)	α (×10 <sup>-4</sup> )	β	
diamond	Egt	5.4125	1.979	- 1437	
Si	$E_{st}$	1.1557	7.021	1108	
Ge	$E_{st}$	0.7412	4.561	210	
Ge	Ead	0.8893	6.842	398	
6H SiC	$E_{gl}$	3.024	0.3055	311	
GaAs	End	1.5216	8.871	572	
InP	E	1.4206	4.906	327	
InAs	Est	0.426	3.158	93	

Est - indirect gap Egd - direct gap

17Y. P. Varshni, Physica 34, 149 (1967).

The bandgap (A versus °K) and the temperature coefficient of the bandgap (A/°K versus °K) of GaAs between 0 and 300°K are shown in the attached figures.



## Ga As (Pankove data)

T(OK)	Eg (eV)	Eg ( Å )	dEg (eV)	dEg (ik)
0	1.5216	8142.7	0.0 x w-4	0.0
10	1.5214	8143.5	0.3	0.162
20	1.5210	8145.9	0.6	0.315
40	1.5193	8155.2	1.12	0.60
60	1.576	8170.0	1.60	0.86
80	1. 513	8189.6	2.04	(. 1)
100	1.508	8214.0	2.44	1.33
150	1. 494	8293.4	3.30	1.83
200	1.496	8396.4	4.00	2.29
250	1.454	8520	4.57	2.68
300	1.430	8664.1	5.05	3.06
400	1.375	9007	5.80	3.80

ONE SPACE PARK - REDONUD BEACH - DAUFORNIA 90278

#### INTEROFFICE CORRESPONDENCE

78.4351.8-013

R. Watson

CC.

DATE 9 February 1978

SUBJECT Swept Spectrum Temperature Gauge

FROM M. Sharma

BLDG

MAIL STA.

EXT

RI

1070

61105

#### Problem

- The temperature-dependent bandedge shift in the crystal is the basis of the optical temperature gauge. Measuring this shift is really the key issue in the design of the temperature gauging.
- There are two methods that have been suggested for measuring the shift. One depends upon measuring the transmitted light intensity through the sensing crystal and the other measures the spectrum of the transmitted light using gratings, monochromators, prisms, etc.
- There are several problems associated with these approaches:
  - Extraneous drift
  - Limited temperature range.
  - Transmitted intensity output versus temperature is not a linear function. Some sort of nonlinear scaling factors will be required.
  - Complexity and very low-signal-level electronics required-electronics becomes a critical technology.

### **Objectives**

To study and incorporate methods that will alleviate the above

#### **Approach**

- The new approach draws an analogy from the electronics measurement system where a bandpass of electronic filters can be measured by a sweep frequency generator.
- In the case of the optical temperature gauge, the spectral content of the light source is made to vary as a function of time; this is equivalent to a sweep generator. If time were to be measured from a reference to the time when the spectral characteristics of the light source and the sensing crystal match, the time interval would indicate the temperature.

A5-1

Page 2 78.4351.8-013 9 February 1978

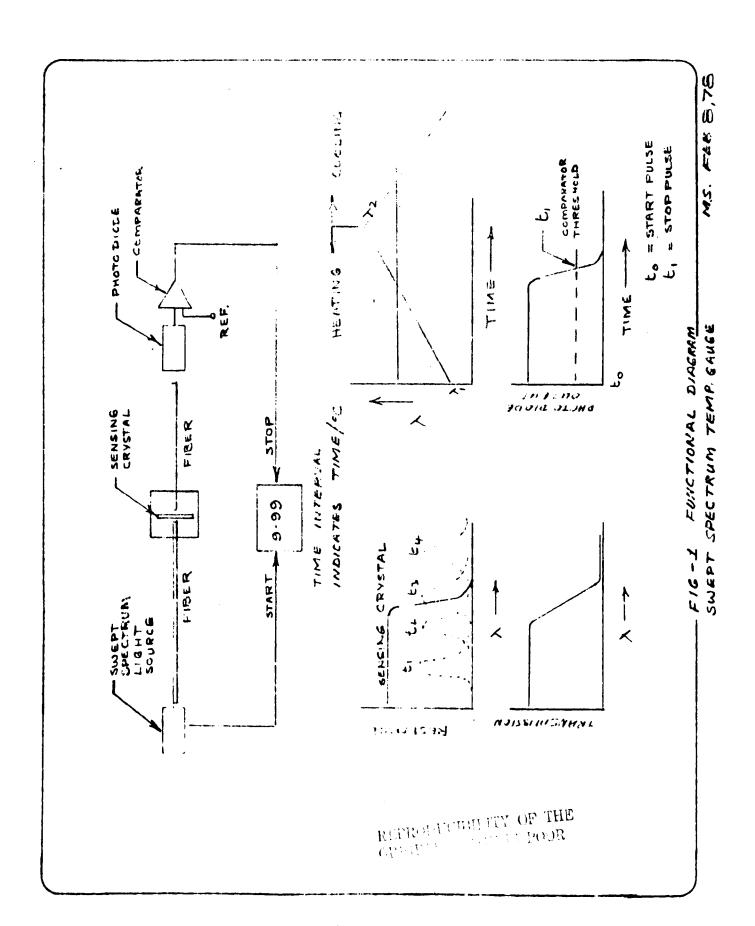
- To employ this concept for temperature gauging, the spectrum of the light source must be varied in time; this can be done several ways:
  - The spectral response of an LED is dependent upon the bandedge; therefore upon the temperature of the LED chip. The chip temperature can be varied by applying power to the LED. To do this, known power is applied to the LED. If the chip is thermally isolated (has no heat sink), its temperature will rise with time linearly. The rate of rise will be a function of the mass of the chip, thermal capacity, and density, and will be controlled by the amount of power applied to it. This approach is a simple means to chirp the spectral output of the LED.
  - A similar approach can be applied to a laser diode since lasing wavelengths are close to the bandedge. The spectral width of laser diodes is very narrow, therefore, a high degree of resolution can be obtained.
  - Similar to the swept LED and the swept laser diode, a crystal can be made to perform the sweeping function. This would be advantageous in the case where the handedge of the LED falls outside the range of measurement or other than LED light source is more desirable.
  - A sweeping monochromator in front of the light source can be employed where high resolution is needed. This monochromator can be simple since it is close to the light source and enough optical power is available.

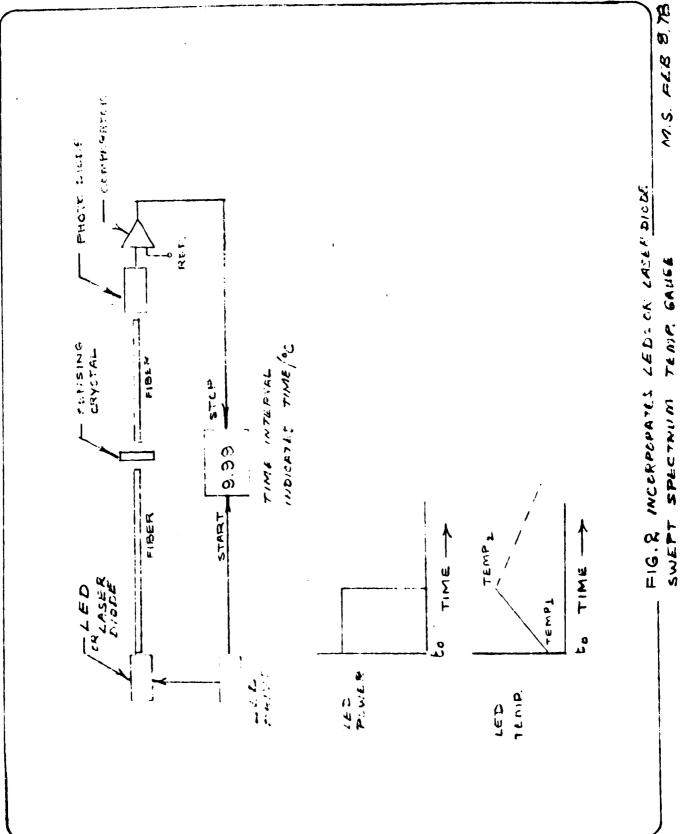
#### **Progress**

- Sketches of 4 conceptual arrangements are shown.
- A dummy temperature gauge has been made for show-and-tell.
- Functional diagram is as shown.
- Pertinent characteristics of candidate crystals have been tabulated.

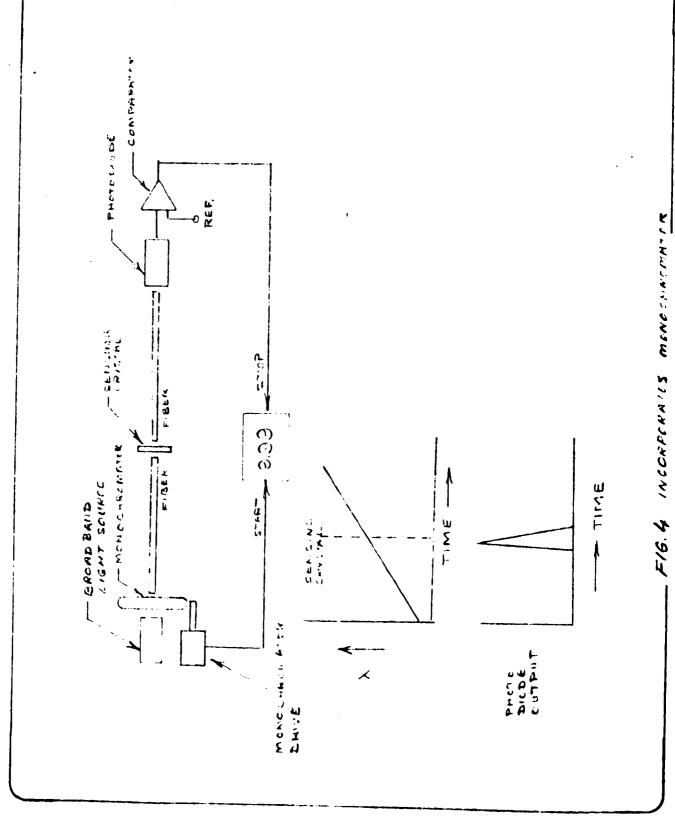
MMS:bj

Attachments





A5-5



f Disparity in classification. Note: Varbers in brackets ([]) are alternate values of the first value given above, obtained from another source. Py \* longitudingl effective mass; mg \* transverse effective mass; mh \* light hole effective mass; mh \* heavy hole effective mass.

Peferers. S. M. Sze. Physics of Sentonductor Davices. John Hiley & Sons, Hew York, 1959.

## PROPERTIES OF SEMICONDUCTORS

	Fnerg E <sub>R</sub> (0 K) eV		Contractor contract	$\begin{pmatrix} dE_R \\ dT \end{pmatrix}$ (300 K) eV K	$\binom{dL_{\mathbf{z}}}{dP}_{T}$ 10°	Effective	mass m*	Refractive index	Static dielectric constant e	Lattice constant a A	Mob μ <sub>e</sub> cm² V+sec	nhty μ <sub>h</sub> em² V-sec
(S)	1.166	1.11	ind 100	2.3	1.5	m, 0.98 m, 0.19	0.52	3,44	11.7	5.43	1.350	480
IV Ge	0.74	0.67	ind 111	- 3.7	5.0	m, 1.58	0.3	4.00	16.3	5.66	3,900	1,900
	0.21		dir 900	<del> </del>	5.0	0.02				6.489	2,000	1,000
la-Sn	3.0	2.8 - 3.2†	ind	-3.3				2.69 c 2.65 c	10.2	a 3.0817 c 15.1123	400	
IV-IV SiC {a	(6H) 2.68	2.2	ind	<del></del>		1				4.359	ļ	
{ Se	1.95	1.74	dir 0001	- 14	- 20		0.12	5.56 c 3.72 (c	8.5		1	
VI Te	0.334	0.32	dir 0001	0.3	19	0.038 ;	0.26 0.10 t	3.07 c 2.68 ; c	5.0 % c 2.2 i c		1,100	

			2 1	ind	1	1	1	ŧ	2.6	6.9	4.538		
	AIP	2.5	-241	ind 100	- 3.5		0.13		3.0	9.8	5.462	80	
İ	AlAs	2.24	2.16	ind			0.5	$\frac{m_l}{m_r} \frac{1.06}{0.49}$		12	5.66	1,000	~100
	1 L		1.6	ind 100	4	1.6		0 19	3.4	11	6.135	50	400
	AISb	1.6		dir 0000	3.9	3.7			2.4		a 3.18 c 5.16	150	
	GaN	3.6	3.5		l.				3.37	10	3 450	120	120
111-V	\GaP	2.4	2 25	ind 100	3.4	1.7	0.13	0.8‡	1.17		3651	8,600	400
_	GaAs [	1.520	1.43	dir 000	5.0	11	0.07	0.3	- 13	15	6.095	4.000	630
<u> </u>	GaSb	0.81	0.69	dir 000	4.1	12	0.045	0.19		121	5 8687	4.000	650
-	InP [	1.42	1.28	dir 000	4.6	4.6	0.07	0.40	3.42	125	6.058	70,000	240
	InAs	041	0 16	dir 000	3.3	5	0.028						3,000
	InSb	0.235	0.17	dir 000	2.9	15	0.0133	0.18	3.75	18	6.4787	76,000	(78°K)
	(ZnO		3.2	dir 0000	-9.5	0.6	0.32	0.27	2.2	7.9	a 3.2496 c 5.2065	180	
	ZnS (a		3.8	dir 0000	- 3.8	9	0.28	0.5	2.4	8.3	a 3.814 c 6.257		
	1		3.6	dir 000	-33	3.7	0.19		2.4	8.3	3,406		
	ZnSe	2 80	2 58	dir 000	7.2	6	0.17		2 89	8 1	5.667	100	
	ZnTe	7 10	2 28	dir 000	-3	-6	0.15		3.36	97	6 101		7
11-VI		2.58	2.53	dir 0000	- 5	3.3	0.20	0.7 ( c	2.5	8 4	a 4 136 c 6.713	210	
	CdSe	1.85	1.74	dir 0000	4.6		0.13	25 1	<del></del>	10.6	a 4 299 c 7.010	500	
		100	1 50	dir 000	41	1.3	0.11	<del>-</del> 6 75	775	103	6 477	600	
	CdTe	1 60	2.30	uir (AA)		1.3				ļ	·	1	
	HgS	- 0.24	013	dir 000			0 043			23	6.083	3,3(x)	
	HgSe		015			·			1.7	20	0.42	22,000	100
	HigTe	- 0.28	0.14	dir 000	- 5.6		0.029	s 0, 1	i e	1		1	(20 K)
	PbS	0.29	0 17	dir III	. 4	7	0 1	01	1.7	170	3 916	330	
	PbSe	0.15	0.26	dir III	- 4	8	$m_i = 0.07$ $m_i = 0.019$	m <sub>e</sub> 0.06		250	6 124	1,020	930
IV-VI	PbTe	0.19	0 29	dir 111	1.4	. 9	m, 0 24 m, 0 02	m, 0. V	1.8	412	6 460	1,620	750
	SnTe	0.3	0 18	Jir 111	<del> </del>	<del> </del>	<u> </u>	1	<del> </del>		6 128		
	(31116	V)	0.10	J 411 111			<del></del>						

The data was gathered from many sources, primarily the following: C. Benoit-à-la-Guillaume et al., Selected Constants Relative to Somiconductors, Pergamon (1961); Solids Under Pressure, ed. W. Paul and D. M. Warschauer, McGraw-Hill (1963), p. 226, J. Tauc, Progress in Semiconductors, 9, 120 (1965); D. Long, Energy Bands in Semiconductors, Wiley (1968); S. Shionoya, "Luminescence of Lattices of the 7ns Type," Luminescence of Organic Solids, ed. P. Goldberg, Academic Press (1966), p. 206; G. Giosecke, "Lattice Constants," Semiconductors and Semimetals, Academic Press 2, 73 (1966); and a number of more recent papers. Where conflicting values appeared, the most recent data was use. Many parameters which depend on the purity of the semiconductor are subject to change as material technology improves.

1Depends on polytype.

1Calculated value.

Calculated value.

table 3-1 Ideal-work requirements for the liquefaction of gases beginning at 70°F and 1 atm

Gas	Normal boiling point, °R	Ideal work of liquefaction —Wi/m, Blu/lb		
Helium, He	7.60		2870	
Hydrogen, H <sub>2</sub>	36.7		5002	
Ncon, Ne	48.8		557.5	
Nitrogen, N <sub>2</sub>	139.2	71°K	· · <del>-</del>	
Air	141.8	18.7	308.6	
Carbon monoxide, CO		81.5	320.7	
Fluorine, Fr		85.4		
Argon, A	157.1	87.5	200.9	
Oxygen, O <sub>2</sub>	162.4	90.2	264.7	
Methane, CH <sub>4</sub>	201.0	112.2		
Carbon tetrafluoride, CF.	261.3	145	73.9	
Ethane, CsHa	332.1		142.4	
Ammonia, NH <sub>2</sub>	431.6		100.2	

#### COPPER vs. COPPER-NICKEL

TYPE T

(Copper-Constantan)
Temperature in Degrees F

Reference Junction at 32°F ...

REPRODUCTION ITY OF THE CAGNAL PAGE IS POOR

DEG F	0	1	2	3	•	`	6	7	•	9	10	Dŧ← F
-190 -180 -170 -160 -150	-1.844 -1.717 -1.565	-3.879	-4.037 -3.844 -3.746 -1.546	-4.051	-3.923 -3.776 -3.676	-4.974 -3.437 -1.741 -1.661	-4.093 -1.951 -1.656 -1.656 -1.563	-4.107 -7.966 -3.675 -3.671	-4.121 -1.980 -1.835 -1.687	-1.494 -1.650 -1.102	-4.147 -4.007 -7.664 -3.711 -1.565	-140 -180 -170 -16J -15U
-140 -130 -120 -110 -100	-3,089 -2,923	-1.767 -3.105 -2.439 -2.771 -2.598	-3.121 -2.956 -2.788	-3.138 -2.973 -2.805	-3.154 -2.089 -2.822	-3.170 -3.006 -2.838	-1,347 -1,186 -1,023 -2,655 -2,685		-3.219	-3.394 -3.235 -3.072 -2.906 -2.736	-1.410 -3.451 -1.089 -2.923 -2.753	-140 -130 -120 -110 -100
-90 -80 -70 -60 -50	-2.405 -2.225 -2.042 -1.856 -1.667	-2.422 -2.243 -2.061 -1.875 -1.686	-2.261	~2.458 -2.279 ~2.098 ~1.912 -1.724	-2.297 -2.116 -1.931	-2.315 -2.134 -1.953	-2.511 -2.333 -2.152 -1.968 -1.761	-7.528 -2.351 -2.171 -1.987 -1.800	-2.169 -2.169 -2.005	-2.563 -2.387 -2.207 -2.024 -1.838	-2.581 -2.405 -2.25 -2.042 -1.856	- 90 - 80 - 70 - 60 - 50
-40 -30 -20 -10	-1.475 -1.279 -1.081 -0.879 -0.674	-1.494 -1.299 -1.101 -0.899 -0.695	-1.319 -1.121 -0.920	-1.533 -1.330 -1.141 -0.940 -0.736	-1.160	-1.377 -1.160 -0.990	-1.591 -1.397 -1.230 -1.000 -0.798	-1.610 -1.416 -1.223 -1.621 -0.818	-1.629 -1.636 -1.243 -1.041 -2.838	-1.648 -1.455 -1.260 -1.061 -2.859	-1.667 -1.475 -1.279 -1.081 -3.879	-40 -30 -20 -10
DEG F	J	2	2	3	•	,		7	•	,	10	DEG F
0 10 20 30 40	-0.674 -0.467 -0.256 -0.043 0.173	-0.654 -0.446 -0.235 -0.022 0.195	-0.633 -0.425 -0.214 -0.000 0.216	-0.613 -0.606 -0.193 0.027 0.238	-1.592 -0.383 -0.171 -0.143 -0.260	-9.362	-0.550 -0.341 -0.129 -0.086 0.303	-0.529 -0.320 -0.107 -0.108 0.325	-0.509 -0.299 -0.096 0.130 0.347	-0.488 -0.277 -0.064 0.151 0.369	-0.467 -0.256 -0.043 0.173 0.391	10 20 30
50 60 70 80 90	0.391 0.611 0.834 1.060 1.288	0.413 0.634 0.857 1.082 	0.439 0.656 0.679 -1.1.5 1.334	0.457 0.678 	0.79 0.779 0.974 1.151 1.380	0.722	0.523 0.745 0.969 1.196 1.426	0.545 0.767 0.992 -1.219 1.449	1 : 24. 1 : 24. 1 : 4.		0.611 0.834 1.060 1.288	50 70 80 90
110 120 130 140	1.752 1.752 1.988 2.226 2.467	1.542 1.775 -2.011 2.250 2.492	2.635 2.274 2.516 شن	1.#27 7.659 2.558 2.550	-1.611 1.846 2.14 -2.177 2.565	1.615 1.819 *2.1:7 7.347 2.589	1.658 1.693 2.131 2.371 -2.613	2.638	1.705 1.666 7.174 2.666		1.752 1.968 2.76 2.467 2.711	100 170 140 140
150 160 170 180 190	2.711 2.958 3.206 3.458 3.711	2.736 2.982 3.231 3.483 3.737	2.760 -3.007 3.256 3.509 3.762	2.785 3.032 3.533 3.533 3.788	3.05/ 3.05/ 3.307 3.559 3.613	2.814 3.082 1.33, 3.584 3.839	2,859 3,107 3,157 3,609 3,864	3.131 3.382 3.635 3.890	-2.908 3.146 3.467 3.660 3.916	2,933 3,181 3,432 3,686 3,941	2.958 3.206 3.458 3.711 3.967	160 170 180 180
200 210 210 210	3.967 4.225 4.486 4.749 5.014	3.993 4.251 4.512 4.775 5.040	4.536	4.765 4.765 4.678 5.093	4.070 4.329 4.591 4.854 5.120	4.096 4.355 4.617 4.881 5.147	4.122 4.581 4.643 4.907 5.174	4.670 4.934 5.200	4.174 4.634 4.696 4.960 5.277	4.199 4.560 4.722 4.987 5.254	4.225 4.486 4.749 5.014 5.281	210 220 230 240
250 260 270 280 290	5.281 5.550 5.821 6.094 6.369	5.307 5.577 5.848 6.122 6.377	, . p / 5	5.361 5.631 5.43 6.177 6.452	5.398 5.658 5.939 6.206 6.480	5.415 5.685 5.957 6.232 6.508	5.442	5.469 5.719 6.612 6.287 6.563	6.4767 6.767 6.734 6.714 6.571	5.523 5.794 6.967 6.342 6.619	5.553 5.921 6.094 6.369 6.647	250 260 270 280 290
310 320 330 340	6.926 7.207 7.490 7.175	6.954	6.782 763 767 7.032	6, '3-) 7, 03 7 7, 292 7, 575 7, 861	F. 758 7. 118 7. 1 U 7.6 % 7.88 #	6,785 7,056 7,348 7,632 7,918	6.414 7.14 7.17 7.561 7.947	7.689	6.8°C 7.1°1 7.6°3 7.7(8 6.004	6.998 7.179 7.462 7.746 8.033	7.207 7.490 7.775 8.062	300 310 320 330 340
350 360 370 380 390	8.062 8.350 8.641 8.933 9.227	8.379 8.670 8.962	A.45 A A.664 B.647	8.148 8.437 8.728 9.021 9.316	8.177 8.466 8.757 9.750 9.145	8.206 8.495 8.787 9.787 9.375	8,215 8,524 8,816 9,107 9,404	8.553	8.574 9.168 9.664	8.321 8.612 8.404 9.198 9.443	0.150 0.641 0.933 9.227 9.523	350 350 370 380 390
400 410 430 440	9.523 9.820 10.120 10.420 10.723	9.850 10.150 10.451 13.753	10.100 10.481 10.784	10.210	10,740	9.671 9.970 10.270 10.572 17.875	9.701 10.000 10.400 10.400 10.900	10.416	10.966 10.966	9.791 10.090 10.390 10.693 10.997	10.120 10.420 10.723 11.027	400 410 420 430 440
450 460 470 460 490	11.027 11.333 11.640 11.949 12.260	11.50 11.67 11.98 17.29	11.194	11,425 11,713 12,142 12,353	17.071	11.775 11.775 12.134 12.415	11,876	12.478	11.587 12.199 12.5)9	11.302 11.610 11.918 12.229 12.540	11.640 11.640 11.949 12.760 12.572	4 50 4 70 4 70 4 40
500 510 520 550 560	12,572 12,885 13,200 13,516 13,834	12.917 13.232 13.548	12.948 13.263 13.580 13.898	11,479 11,795 11,611 13,930	13, 111 13, 176 13,643 13,961	19.993	14.025	13.105	11.117 13.453 11.770 14.059	14.1/1	14.151	500 510 520 530 540
590 570 570 580	14.153 18.474 14.795 15.116 15.443	14.506	14.514	14,570	14,602	14.513 14.634 14.557 15.28 15.605	14, 145 14,666 14,949 15,111 15,618	14.699 15.021 15.345	14.731 15.554 15.379	14.761 15.086	14.795	5 50 5 60 5 70 5 80 5 90
630 610 630 630	19.769 16.096 16.424 16.791 17.084	16.175	16.141 7 14.47 5 16.41	16.194	16.727 16.555 16.555	16.759 16.598 11.414	16.471	16.124 16.654 16.654	17.35# 16.687 17.31#	16.770	16.424	610 610 630 630
650 660 670 660	17.416 17.750 18.486 18.420 18.75	17.70' 6 10.115 7 10.45	1 17.816 8 18.151 6 18.487 1 18.824	17.85 1 18.185 7 18.571 18.858	17.84) 18.218 18.555 18.655	17,717 18,257 18,568 18,426	10.05	17.79 18.119 18.55 18.491	18.017 18.353 18.494 19.027	19.051	18.084 18.470 18.757	61) 640 670 680 690
740 710 720 730 740	19.0 25 19.436 19.776 20.116 20.436	19.441 19.801 20.150	19.40.0 19.44.0 20.104 20.526	19.494 19.471 20.216 20.560	19,410 19,411 20,252	14.404	19.616	19.672 20.011 20.355	19.700 20.547 20.499	20.001	20.116	700 710 720 730 740
790		1 20.49	30.465									750
0EG / • ( 0#	v##1€0 #1	NOM DEGM	ees cites	15 19681	. •	•	•	,	•	•	10	066 1



3

COPPER-CONSTANTAN THERMOCOUPLES Degrees Fahrenheit (Int. 1948) Cold End at 75° F  $\,$  (EMF at 32° F = -0.944 mv)

	نو	-300 -200 -100 - 0	F	200 300 400 500 500	500 200 700
	56∸	-6.179 -4.986 -3.416	95	0.457 2.895 5.564 8.432 11.474	14.666
	06-	-6.129 -6.179 -4.916 -4.986 -3.329 -3.416	8	0.342 2.768 5.426 8.285 11.319	14.503
	23	-6.078 -4.845 -3.240	85	0.227 2.640 5.288 8.138 11.164	14.340
	98-	-6.025 -4.773 -3.151	8	0.113 2.514 5.150 7.991 11.009	14.017 14.178
	-75	-5.973 -4.701 -3.061	75	0.000 2.388 5.013 7.844 10.853	
	70	-5.385 -5.448 -5.510 -5.571 -5.632 -5.691 -5.749 -5.807 -5.863 -5.918 -5.973 -6.025 -6.078 -6.129 -6.179 -3.924 -4.006 -4.087 -4.167 -4.245 -4.324 -4.401 -4.477 -4.553 -4.628 -4.701 -4.773 -4.845 -4.916 -4.986 -2.115 -2.214 -2.311 -2.407 -2.503 -2.598 -2.692 -2.786 -2.878 -2.970 -3.061 -3.151 -3.240 -3.329 -3.416	67	2.263 2.263 4.877 7.698 10.699	13.855
	-65	-5.863 -4.553 -2.878	65	2.138 2.138 4.741 7.553 10.546	13.533 13.693 16.81G 16.977
15	-60	-5.807 -4.477 -2.786	3	2.014 2.014 4.606 7.408 10.392	
ILIVOL	55	-5.749 -4.401 -2.692	55	-0.445  .891  4.471  7.263	
JTE MI	-50	-5.691 -4.324 -2.598	8	-0.555 1.767 4.336 7.120 10.086	13.053 13.213 16.311 16.477 19.690 19.861
EMF IN ABSOLUTE MILLIVOLTS	-45	-5.385 -5.448 -5.510 -5.571 -5.632 -5.691 -5.749 -5.807 -5.863 -3.924 -4.006 -4.087 -4.167 -4.245 -4.324 -4.401 -4.477 -4.553 -2.115 -2.214 -2.311 -2.407 -2.503 -2.598 -2.692 -2.786 -2.380	45		
IF IN	9	-5.571 -4.167 -2.407	Ŝ.	-0.880 -0.773 1.402 1.523 3.936 4.070 6.589 6.832 9.530 7.782	12.734 12.894 15.980 16.145 19.347 19.519
<b>E</b>	-35	-5.510 -4.087 -2.311	35	1.402 1.402 3.936 6.589 9.530	12.418 12.576 12.734 12.894 15.549 15.814 15.980 16.145 19.005 19.176 19.347 19.519
	-30	-5.385 -5.448 -3.924 -4.006 -2.115 -2.214	30	-0.986 1.282 3.805 6.547 9.479	12.576 15.814 19.176
	-25	-5.385 -3.924 -2.115	25	1.163 1.163 3.673 6.405 9.329	(5.549 (5.549 (9.005
	-20	مـــما	20		12.259
	-5 -10 -15 -20	-300   -6.228   -6.276   -6.323   -5.256   -5.321   -5.00   -5.256   -5.321   -100   -3.533   -3.84   -3.758   -3.84   -1.00   -3.533   -1.815   -1.917   -2.016	25	: [	11,631 11,788 11,944 12,102 12,259 14,829 14,993 15,157 15,320 15,485 18,156 18,325 18,495 18,664 18,835
	-10	-6.323 -5.190 -3.674 -1.816	2	1.614 —1.511 —1.407 0.633 0.689 0.807 3.023 3.152 3.281 5.703 5.842 5.982 8.58- 8.730 8.879	11.944 15.157 18.495
	ې	-300   -6.228   -6.276   -6.323 -200   -5.055   -5.123   -5.190 -100   -3.503   -3.589   -3.674 0   -1.614   -1.715   -1.816	8	0.573 0.689 3.023 3.152 5.703 5.842 8.58 8.730	11.788
	•	-5.055 -3.503	0	6 -1.614 100 0 573 200 3.023 306   5.763 406   8.58	14.829
	۳	-300 -200 -100		200 200 300 400	500 600 700

(24B) OC

COPPER-CONSTANTAN THERMOCOUPLES Degrees Celsius (Centigrade, Int. 1948) Cold End at 25  $^\circ$  C (EMF at 0  $^\circ$  C = -0.990 mv)

	ڼ	8 °	ڼ	200 300 400
	-95	-4,197	95	3.054 8.033 (3.583 19.575
	06-	-5.004 -5.128 -5.249 -5.367 -5.482 -5.593 -5.702 -5.807 -5.909 -6.008 -6.103 -6.195 -6.284 -6.369 -1.923 -2.102 -2.279 -2.453 -2.794 -2.96! -3.125 -3.286 -3.445 -3.461 -3.754 -3.904 -4.052 -4.197	%	2.823 7.769 13.295 19.267
	-85	-6.284 -3.904	85	2.594 7.507 13.007 18.959
	-80	-6.195	Cg	2.367 7.246 12.720 18.652
	-75	-6.103	7.5	2.142 6.988 12.435 18.347
	-70	-6.008	20	1.918 6.729 12.150 18.042
	60657075 <b>80</b>	-5.004 -5.128 -5.249 -5.367 -5.482 -5.593 -5.702 -5.807 -5.909 -6.008 -6.1036.195 -6.284 -6.369 -1.923 -2.102 -2.179 -2.655 -2.794 -2.96! -3.125 -3.286 -3.445 -3.461 -3.754 -3.794 -4.052	65	1.477 1.697 6.218 6.472 11.585 11.867 17.435 17.737
15	09	-5.807	8	1.477 6.218 11.585 17.435
EMF IN ABSOLUTE MILLIVOLTS	-5055	-5.702	25	1.260 5.964 1.304 7.133
JTE MI	-50	-5.593	Š	1.045 5.713 11.025 16.831
ABSOLL	-35 -4045	-5.482	45	0.831 5.463 10.745 16.53
Z	8	-5.367	\$	
E	-35	-5 249	35	0.411 4.967 10.193 15.934
	-25 -30	-5.128	8	0.204 4.722 9.919 15.636
	-25	-5.004	ສ	0.000 4.479 9.645 15.340
	-20	-4.877	7	-0.203 4.237 9.373 15.045
	-15	-4.747	5	3.997 9.103 14.750
	C 0 -5 -10 -15 -20	0 -0.990 -1.181 -1.370 -1.557 -1.741	°C. 0 5 10 15 20	0 -0.990 -0.797 -0.601 -0.403 -0.203 100 3.287 3.522 3.759 3.997 4.237 200 8.298 8.565 8.833 9.103 9.373 300 13.874 14.165 14.457 14.750 15.045 400 19.884
	7	-4.478	\$	0.990 -0.797 -0.601 3.287 3.522 3.759 8.298 8.565 8.833 3.874 14.165 14.457
	•	-4.339	[ •	0 -0.990 100 3.287 200 8.298 300 13.874 400 19.884
	ڼ	8 0	نې	200

# END

# 

# FILMED

APR 7 1980